

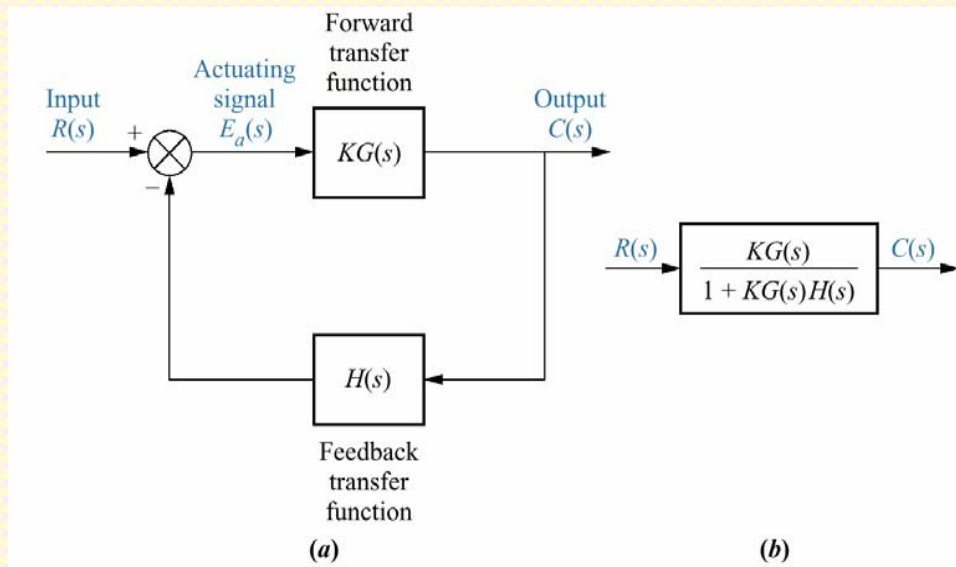
Lecture – 5  
*Classical Control Overview – III*

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# A Fundamental Problem in Control Systems

- Poles of open loop transfer function are easy to find and they do not change with gain variation either.
- Poles of closed loop transfer function, which dictate stability characteristics, are more difficult to find and change with gain



*open loop transfer function =  $KG(s)H(s)$*

*closed loop transfer function*

$$T(s) = \frac{KG(s)}{1 + KG(s)H(s)}$$

*let*

$$G(s) = \frac{N_G(s)}{D_G(s)} \quad \text{and} \quad H(s) = \frac{N_H(s)}{D_H(s)}$$

$$T(s) = \frac{KN_G(s)D_H(s)}{D_G(s)D_H(s) + KN_G(s)N_H(s)}$$

## Example

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$$\text{Let } G(s) = \frac{(s+1)}{s(s+2)}, \quad H(s) = \frac{(s+3)}{(s+4)}$$

$$\text{Then } KG(s)H(s) = \frac{K(s+1)(s+3)}{s(s+2)(s+4)}$$

Poles of open loop Transfer Function  $KG(s)H(s)$  are  $(0, -2, -4)$

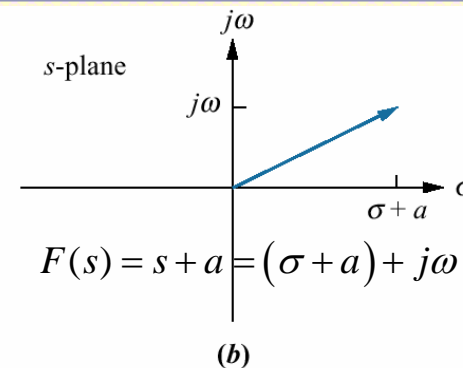
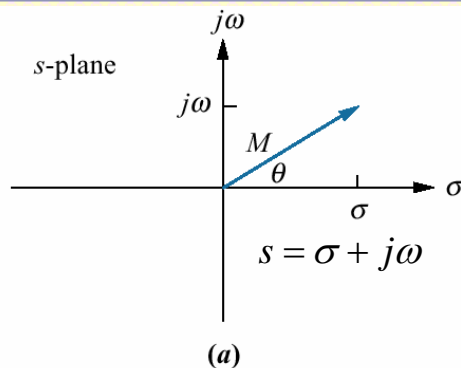
Now Closed Loop Transfer Function  $T(s)$

$$T(s) = \frac{K(s+1)(s+4)}{s^3 + (6+K)s^2 + (8+4K)s + 3K}$$

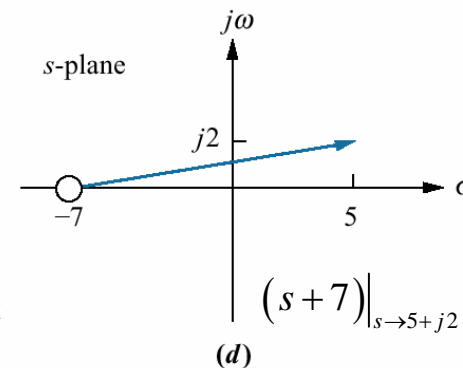
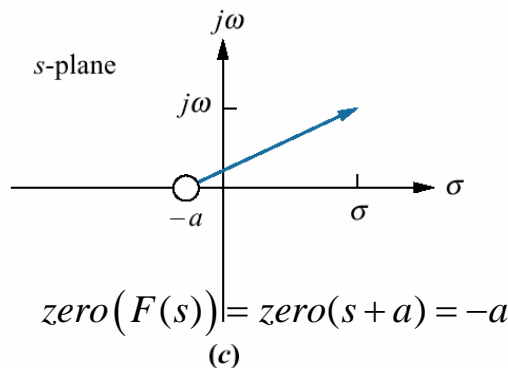
- Poles of  $T(s)$  are not immediately known; they depends on  $K$  as well.
- System stability and transient response depends on poles of  $T(s)$
- Root Locus gives vivid picture of the poles of  $T(s)$  as  $K$  varies

# Vector Representation of Complex Numbers and Complex Functions

Regular representation:



Alternate representation:



We can conclude that  $(s + a)$  is a complex number and can be represented by a vector drawn from the zero of the function  $(-a)$  to the point 's'.

# Magnitude and Angle of Complex Functions

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$$\text{Let } F(s) = \frac{\prod_{i=1}^m (s + z_i)}{\prod_{j=1}^n (s + p_j)}, \quad m - \text{number of zeros, } n - \text{number of poles}$$

The magnitude  $M$  of  $F(s)$  at any point  $s$  is

$$M = \frac{\prod \text{zero lengths}}{\prod \text{pole lengths}} = \frac{\prod_{i=1}^m |(s + z_i)|}{\prod_{j=1}^n |(s + p_j)|}$$

The angle,  $\theta$ , of  $F(s)$  at any point  $s$  is

$$\theta = \sum \text{zero angles} - \sum \text{pole angles} = \sum_{i=1}^m \angle(s + z_i) - \sum_{j=1}^n \angle(s + p_j)$$

# Example

$$F(s) = \frac{(s+1)}{s(s+2)}$$

Vector originating at  $-1$ :  $\sqrt{20} \angle 116.6^\circ$

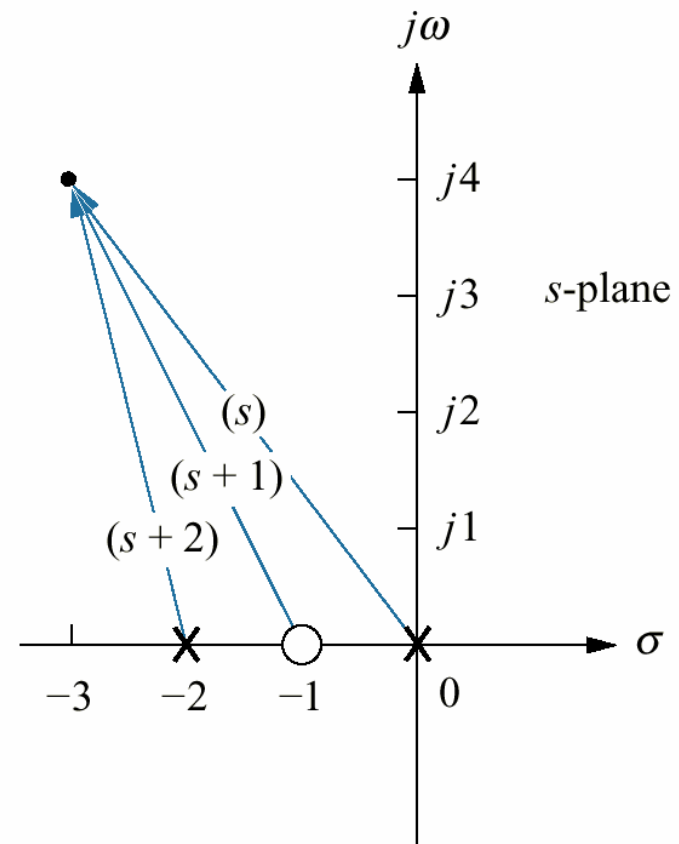
Vector originating at  $0$ :  $5 \angle 126.9^\circ$

Vector originating at  $-2$ :  $\sqrt{17} \angle 104.0^\circ$

$\therefore M \angle \theta$  of  $F(s)$

$$= \frac{\sqrt{20}}{5\sqrt{17}} \angle 116.6^\circ - 126.9^\circ - 104.0^\circ$$

$$= 0.217 \angle -114.3^\circ$$



# *Root Locus Analysis*

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# Facts

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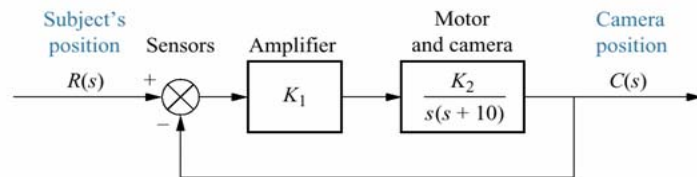
- Introduced by W. R. Evans (1948)
- Graphical representation of the **closed loop poles** in s-plane as the system parameter (typically controller gain) is varied.
- Gives a graphic representation of a system's stability characteristics
- Contains both qualitative and quantitative information
- Holds good for higher order systems



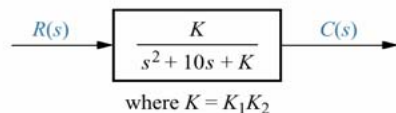
# A Motivating Example



(a)



(b)



where  $K = K_1 K_2$

(c)

**Ref:** N. S. Nise:  
Control Systems Engineering,  
4<sup>th</sup> Ed., Wiley, 2004

**Pole location as a function of gain for the system.**

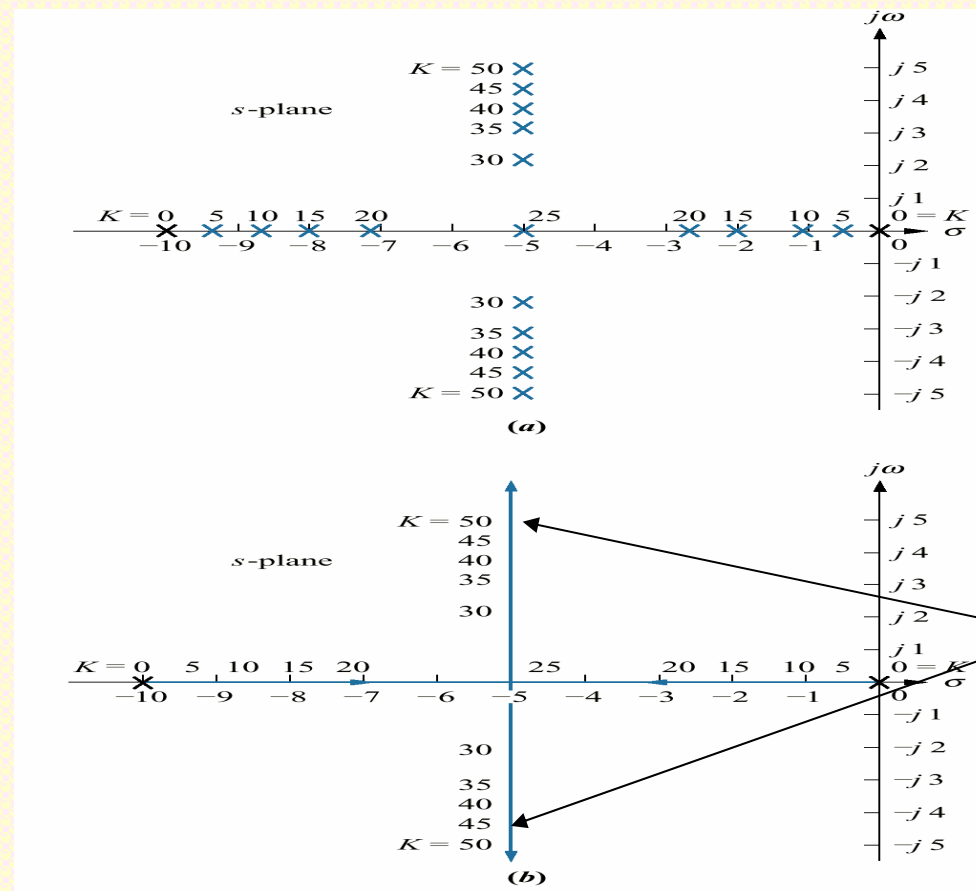
K	Pole 1	Pole 2
0	-10	0
5	-9.47	-0.53
10	-8.87	-1.13
15	-8.16	-1.84
20	-7.24	-2.76
25	-5	-5
30	-5 + j2.24	-5 - j2.24
35	-5 + j3.16	-5 - j3.16
40	-5 + j3.87	-5 - j3.87
45	-5 + j4.47	-5 - j4.47
50	-5 + j5	-5 - j5

over damped

under damped

**Note:** Typically  $K > 0$  for negative feedback systems

# A Motivating Example



Poles are always negative:  
 • system is always stable

Real parts are same:  
 • same settling time

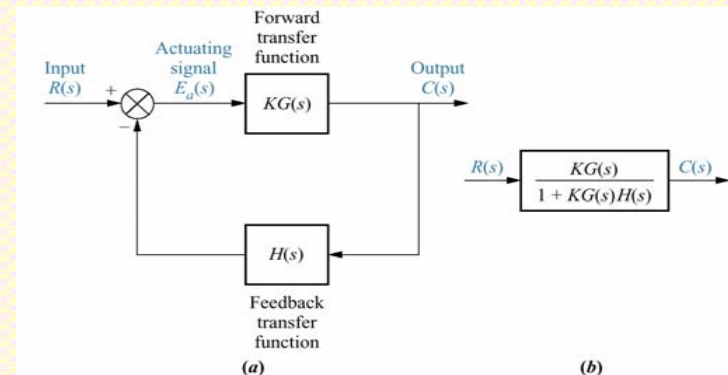
As gain increases,  
 • damping ratio decreases  
 • %age overshoot increases

# Properties of Root Locus

The closed loop Transfer Function

$$T(s) = \frac{KG(s)}{1 + KG(s)H(s)}$$

Characteristics equation:  $1 + KG(s)H(s) = 0$



Closed loop poles are solution of characteristics equation.

However,  $1 + KG(s)H(s) = 0$  is a complex quantity. Hence, it can be expressed as

$$KG(s)H(s) = -1 = 1 \angle (2k + 1)180^\circ \quad k = 0, \pm 1, \pm 2, \pm 3, \dots$$

The above condition (Evan's condition) can be written as

$$|KG(s)H(s)| = 1 \quad (\text{Magnitude criterion})$$

$$\angle KG(s)H(s) = (2k + 1)180^\circ \quad (\text{Angle criterion})$$

# Example

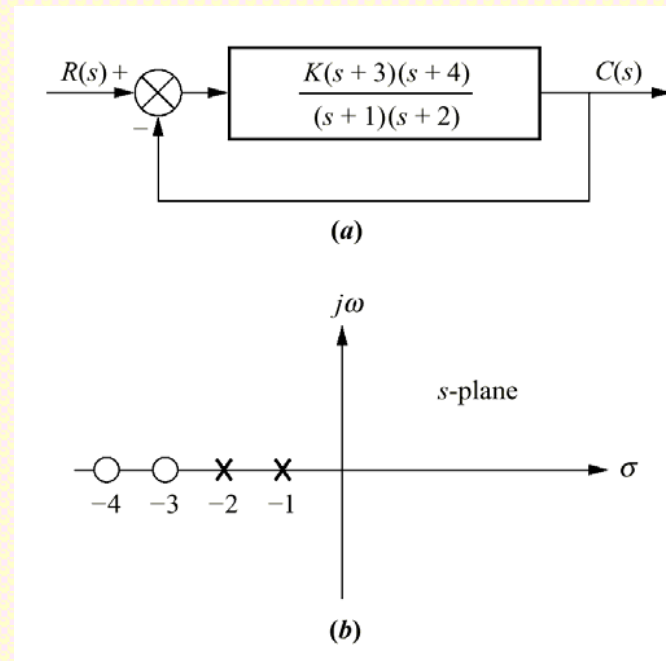
**Fact :** If the angle of a complex number is an odd multiple of  $180^0$  for an open loop transfer function, then it is a pole of the closed loop system with

$$K = \frac{1}{|G(s)H(s)|} = \frac{1}{|G(s)||H(s)|}$$

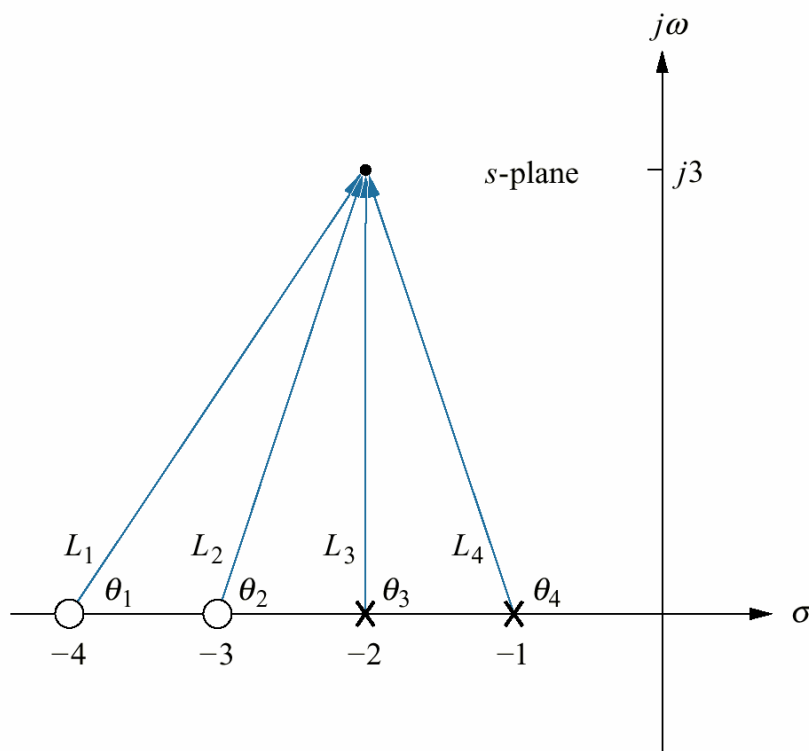
**Example:** Let us consider a system as in the Figure and consider two points:

$$P_1 : -2 + j3$$

$$P_2 : -2 + j\left(\frac{\sqrt{2}}{2}\right)$$



# Example



For  $P_1$ :  $\sum \theta_i = \theta_1 + \theta_2 - \theta_3 - \theta_4 = 56.31 + 71.57 - 90 - 108.43$   
 $= -70.55^\circ \neq (2k+1)180^\circ$

Therefore  $-2 + j3$  is not a point on root locus

For  $P_2$ :  $\sum \theta_i = 180^\circ$

Hence  $-2 + j\left(\frac{\sqrt{2}}{2}\right)$  is on root locus for some value of  $K$ .

Moreover,  $K = \frac{L_3 L_4}{L_1 L_2} = \frac{\frac{\sqrt{2}}{2} \times (1.22)}{(2.12) \times (1.22)} = 0.33$

# Summary

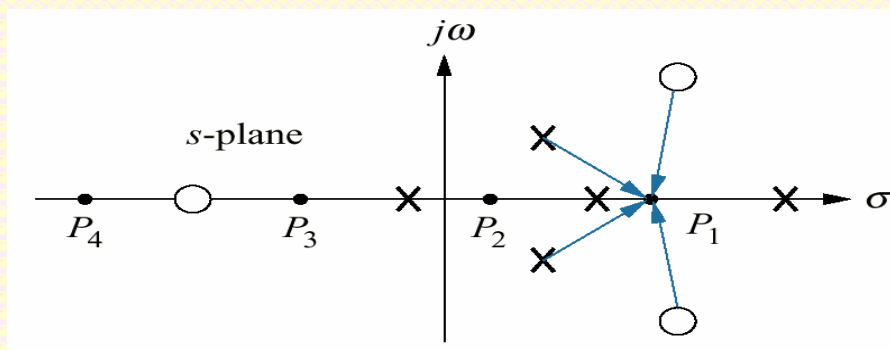
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Given the poles and zeros of the open loop transfer function  $KG(s)H(s)$ , a point in the  $s$  - plane is on the root locus for a particular value of gain  $K$ , if the angles of the zeros minus the angles of the poles add up to  $(2k + 1)180^\circ$ .

Furthermore, the gain  $K$  at that point can be found by dividing the product of the lengths of the poles by the product of the lengths of the zeros.

# Sketching the Root Locus: Basic Five Rules

- 1. Number of Branches:** The number of branches of the root locus equals the number of closed loop poles (since each pole should move as the gain varies).
- 2. Symmetry:** A root locus is always symmetric about the real axis, since complex poles must always appear in conjugate pairs.
- 3. Real-axis segments:** On the real axis, for  $K > 0$ , the root locus exists to the left of an odd number of real-axis finite open-loop poles and zeros.



# Sketching the Root Locus: Basic Five Rules

- 4. Starting and ending points:** The root locus begins at the finite and infinite poles of  $G(s)H(s)$  and ends at the finite and infinite zeros of  $G(s)H(s)$
- 5. Behavior at Infinity:** The root locus approaches straight lines as asymptotes as the locus approaches infinity

The equation of asymptotes is given by the real-axis intercept  $\sigma_a$  and angle  $\theta_a$  as follows:

$$\sigma_a = \frac{\sum \text{finite poles} - \sum \text{finite zeros}}{\text{No. of finite poles} - \text{No. of finite zeros}}$$
$$\theta_a = \frac{(2k+1)\Pi}{\text{No. of finite poles} - \text{No. of finite zeros}}$$

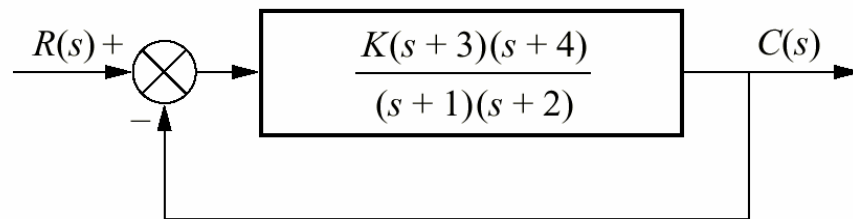
where  $k = 0, \pm 1, \pm 2, \pm 3, \dots$

and the angle is given in radians wrt. positive extension of real axis

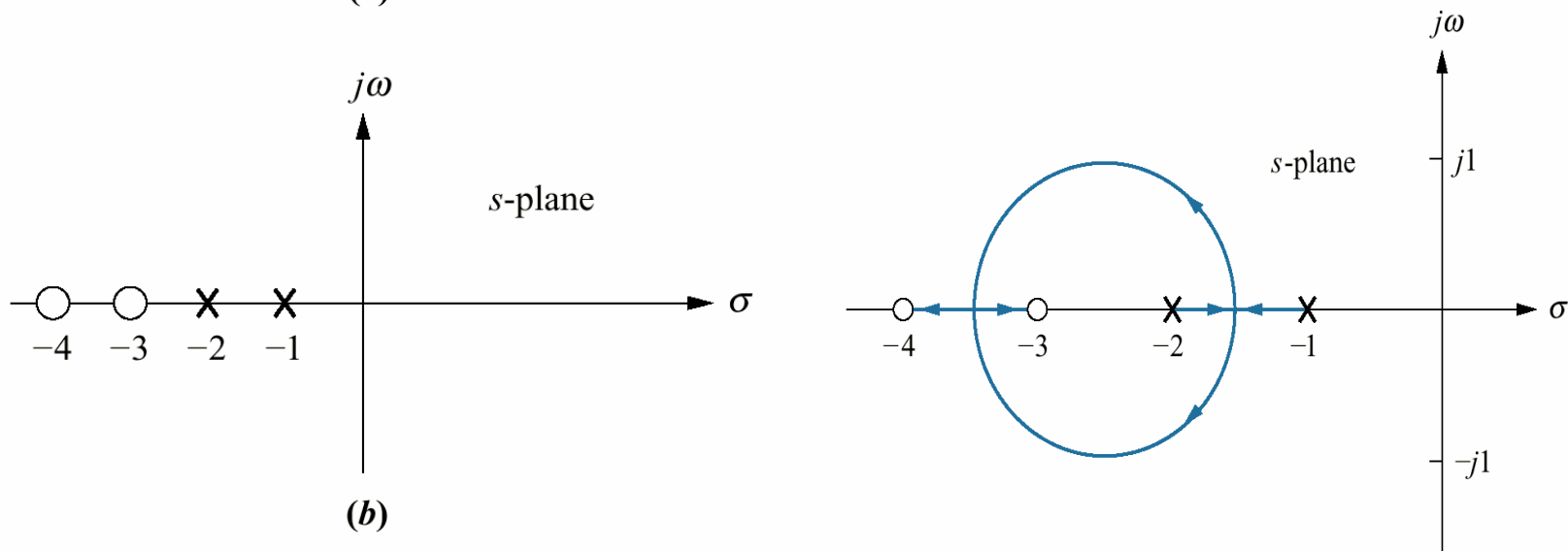
**For additional rules, refer to:**  
N. S. Nise: Control Systems  
Engineering, 4<sup>th</sup> Ed., Wiley, 2004.



# Example

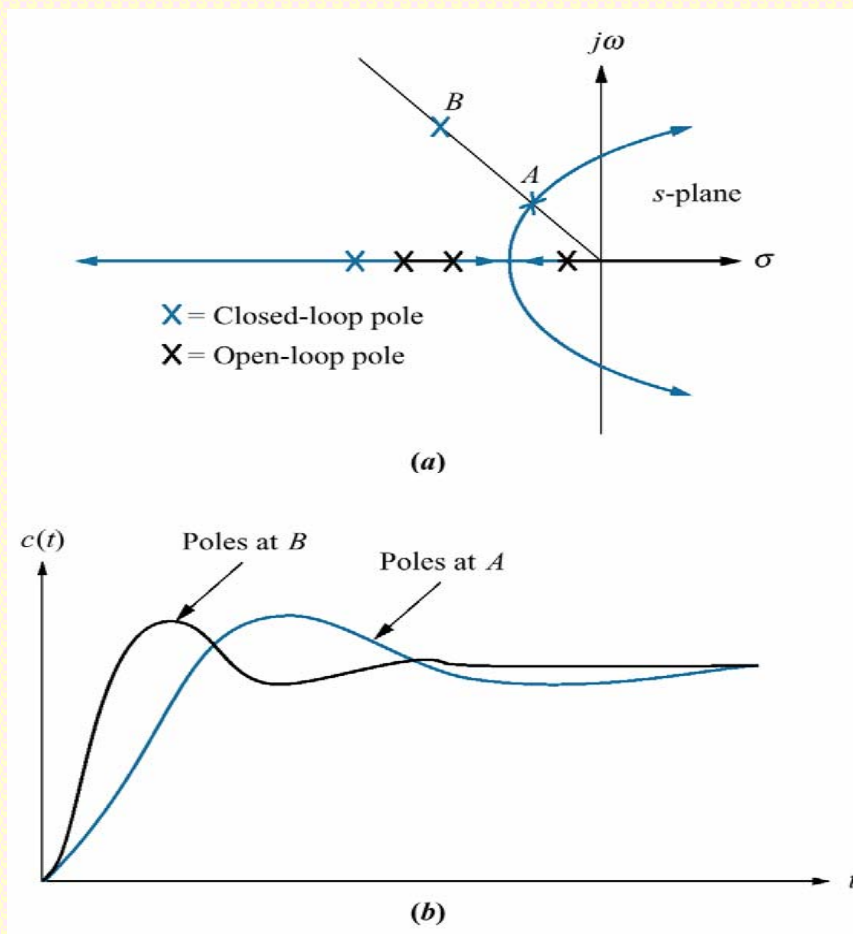


(a)



(b)

# Control Tuning via Gain Adjustment using Root Locus



## Sample root locus:

- A. Possible design point via gain adjustment
- B. Desired design point that cannot be met via simple gain adjustment (needs dynamic compensators)

# *PID Control Design*

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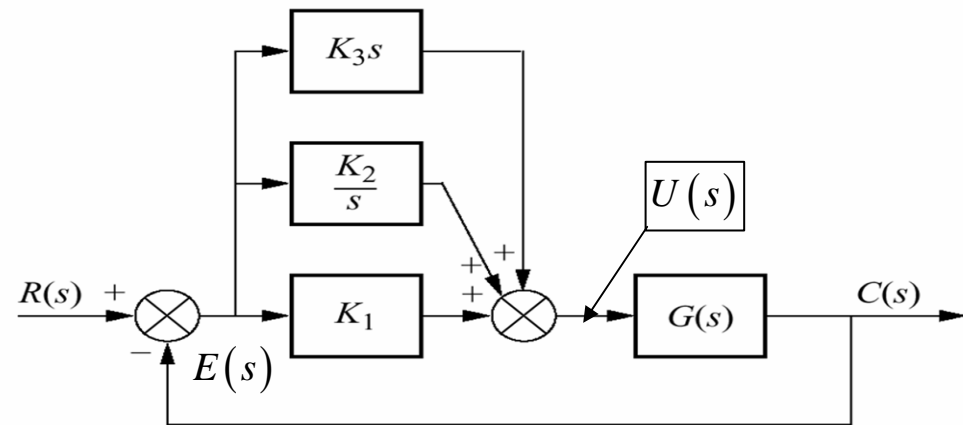
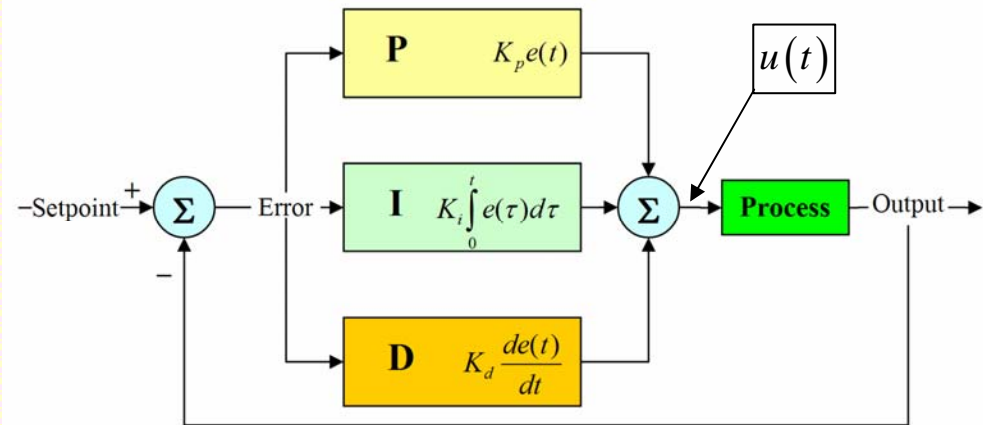
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# Introduction

Ref : [en.wikipedia.org/wiki/PID\\_controller](http://en.wikipedia.org/wiki/PID_controller)

- The PID controller involves three components:
  - Proportional feedback
  - Integral feedback
  - Derivative feedback
- By tuning the three components (gains), a suitable control action is generated that leads to desirable closed loop response of the output.



Ref: N. S. Nise: Control Systems Engineering, 4<sup>th</sup> Ed., Wiley, 2004.

# Philosophy of PID Design

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- The **proportional** component determines the reaction to the **current value of the output error**. It serves as a “all pass” block.
- The **integral** component determines the reaction based on the **integral (sum) of recent errors**. In a way, it accounts for the history of the error and serves as a “low pass” block.
- The **derivative** component determines the reaction based on the **rate of change of the error**. In a way, it accounts for the future value of the error and serves as a “high pass” block.

# PID Controller

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The final form of the PID algorithm is:

$$u(t) = K_p e(t) + K_i \int_0^t e(\tau) d\tau + K_d \frac{d}{dt} e(t)$$

$$\frac{U(s)}{E(s)} = K_p + K_i / s + K_d s$$

The tuning parameters are:

Proportional gain,  $K_p$

Integral gain,  $K_i = K_p / T_i$

Derivative gain,  $K_d = K_p T_d$

# Effect of Proportional Term

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- **Proportional Term:**

$$P_{out} = K_p e(t)$$

$P_{out}$ : Proportional term of output

$K_p$ : Proportional gain (a tuning parameter)

- If the gain  $K_p$  is **low**, then control action may be too small when responding to system disturbances. Hence, it need not lead to desirable performance.
- If the proportional gain  $K_p$  is **too high**, the system can become unstable. It may also lead to noise amplification

# Effect of Integral Term

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## Integral term :

$$I_{out} = K_i \int_0^t e(\tau) d\tau$$

$I_{out}$ : Integral term of output

$K_i$ : Integral gain, a tuning parameter

The integral term (when added to the proportional term) accelerates the movement of the process towards setpoint and eliminates the residual steady-state error that occurs with a proportional only controller.



# Effect of Integral Term

## Destabilizing effects of the Integral term :

It can be seen that adding an integral term to a pure proportional term increases the gain by a factor of

$$\left|1 + \frac{1}{j\omega T_i}\right| = \sqrt{1 + \frac{1}{\omega^2 T_i^2}} > 1, \text{ for all } \omega.$$

and simultaneously increases the phase-lag since

$$\angle\left(1 + \frac{1}{j\omega T_i}\right) = \tan^{-1}\left(\frac{-1}{\omega T_i}\right) < 0 \text{ for all } \omega.$$

Because of this, both the gain margin (GM) and phase margin (PM) are reduced, and the closed-loop system becomes more oscillatory and potentially unstable.

**Ref :** Li, Y. , Ang, K.H. and Chong, G.C.Y. (2006)  
PID Control System Analysis and Design.  
*IEEE Control Systems Magazine* 26(1):pp. 32-41.

# Effect of Integral Term

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- **Integrator Windup:**

If the actuator that realizes control action has saturated and if it is neglected, this causes low frequency oscillations and leads to instability.

**Remedies:**

Automatic Resetting, Explicit Anti-windup.

# Effect of Derivative Term

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## Derivative term :

$$D_{out} = K_d \frac{d}{dt} e(t)$$

$K_d$ : Derivative gain (a tuning parameter)

- The derivative term speeds up the transient behaviour. In general, it has negligible effect on the steady state performance (for step inputs, the effect on steady state response is zero).
- Differentiation of a signal amplifies noise. Hence, this term in the controller is highly sensitive to noise in the error term! Because of this, the derivative compensation should be used with care.

# Effect of Derivative Term

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Adding a derivative term to pure proportional term reduces the phase lag by

$$\angle(1 + j\omega T_d) = \tan^{-1}\left(\frac{\omega T_d}{1}\right) \in [0, \pi/2] \text{ for all } \omega$$

which tends to increase the Phase Margin.

In the meantime, however, the gain increases by a factor of

$$|1 + j\omega T_d| = \sqrt{1 + \omega^2 T_d^2} > 1, \text{ for all } \omega.$$

which decreases the Gain Margin.

Hence the overall stability may be improved or degraded.

**Additional requirement** : Low pass filter, Set-point filter, Pre-filter etc.

# Effect of Increasing Gains on Output Performance

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<b>Parameter</b>	<b>Rise time</b>	<b>Overshoot</b>	<b>Settling time</b>	<b>Error at equilibrium</b>
$K_p$	Decreases	Increases	Small change	Decreases
$K_i$	Decreases	Increases	Increases	Eliminated
$K_d$	Indefinite (can either decrease or increase)	Decreases	Decreases	No Effect

# Tuning of PID Design

<b>Method</b>	<b>Advantages</b>	<b>Disadvantages</b>
<b>Manual Tuning</b>	No math required. Largely trial-and-error approach (based on general observations)	Requires experienced personnel
<b>Cohen-Coon</b>	Results in good tuning in general	Offline method. Good for first-order processes
<b>Ziegler–Nichols</b>	Proven online method.	Very aggressive tuning, May upset some inherent advantages of the process
<b>Software Tools</b>	Online or offline methods.	Some cost and training of personnel is involved

# Ziegler–Nichols Method for Gain Tuning

The  $K_i$  and  $K_d$  gains are first set to zero. The  $P$  gain is increased until it reaches the critical gain,  $K_c$ , at which the output of the loop starts to oscillate. Next,  $K_c$  and the oscillation period  $P_c$  are used to set the gains as per the following rule.

<b>Control Type</b>	$K_p$	$K_i$	$K_d$
$P$	$0.50K_c$	-	-
$PI$	$0.45K_c$	$1.2K_p / P_c$	-
$PID$	$0.60K_c$	$2K_p / P_c$	$K_p P_c / 8$

**Note:** The constants used may vary depending on the application.

# Limitations of PID control

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- It is a SISO design approach and hence can effectively handle only such system
- System should behave in a fairly linear manner. Hence, it is valid in close proximity of an operating point (about which the linearized system is valid)
- Does not take into account the limitations of the actuators

## Techniques to overcome:

- Gain scheduling, Cascading controllers, Filters in loop etc.



**Thanks for the Attention...!**



