Lecture – 28

Linear Quadratic Regulator (LQR) Design – II

Dr. Radhakant Padhi

Asst. Professor Dept. of Aerospace Engineering Indian Institute of Science - Bangalore

Outline

- \bullet Stability and Robustness properties of LQR
- \bullet Optimum value of the cost function
- \bullet Extension of LQR design
	- \bullet For cross-product term in cost function
	- \bullet Rate of state minimization
	- \bullet Rate of control minimization
	- \bullet LQR design with prescribed degree of stability
- **LQR for command tracking**
- \bullet LQR for inhomogeneous systems

Stability and Robustness Properties

Dr. Radhakant Padhi Asst. Professor Dept. of Aerospace Engineering Indian Institute of Science - Bangalore

LQR Design: Stability of Closed Loop System

 \bullet Closed loop system **• Lyapunov function** $\dot{X} = AX + BU = (A-BK)X$ $V(X) = X^T P X$ $=\left[\left(A-BK \right)X \right]^{T} PX + X^{T}P \left[\left(A-BK \right)X \right]$ $\left(A-BR^{-1}B^TP\right)^{\prime}\ P+P\left(A-BR^{-1}B^TP\right)$ $= X^T \left[\left(PA + A^T P - PBR^{-1}B^T P + Q \right) - Q - PBR^{-1}B^T P \right] X$ $\mathbf{Z} = \mathbf{X}^T \mathbf{\left| \right.} - \mathbf{Q} - \mathbf{PBR^{-1}B^TP} \mathbf{\left. \right|} \mathbf{X}$ $\dot{Y} = \dot{X}^T P X + X^T P X$ X^T $(A - BR^{-1}B^TP)^T$ $P + P(A - BR^{-1}B^TP)$ X $= X^T \left[\left(A - RR^{-1}R^T P \right)^T P + P \left(A - RR^{-1}R^T P \right) \right]$ $\left[\left(A-BK\right)X\right]^{T}PX+X^{T}P\left[\left(A-BK\right)X\right]$ $\left[\begin{array}{cc} (A-BR^{-1}B^{T}P) & P+P(A-BR^{-1}B^{T}P) \end{array} \right]$ $\left[\left(PA + A^TP - PBR^{-1}B^TP + Q\right) - Q - PBR^{-1}B^TP\right]$ $\left[-Q-PBR^{-1}B^{T}P\right]$ $\dot{V} = \dot{Y}^T P Y \perp Y^T P \dot{Y}$

LQR Design: Stability of Closed Loop System

 $Hence, (PBR^{-1}B^{T}P + Q) > 0$ $For R > 0, R^{-1} > 0. Also P > 0$ S ^o $PBR^{-1}B^{T}P>0$ Also $Q \geq 0$. > 0 , $R^{-1} > 0$. Also $P >$

 \therefore $\dot{V}(X) < 0$ Hence, the closed loop system is always asymptotically stable!

LQR Design: Minimum value of cost function

$$
J = \frac{1}{2} \int_{t_0}^{\infty} (X^T Q X + U^T R U) dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + \left(-R^{-1} B^T P X \right)^T R \left(-R^{-1} B^T P X \right) \right] dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} X^T \left(Q + P B R^{-1} B^T P \right) X dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} (-V) dt = -\frac{1}{2} \left[V \right]_{t_0}^{\infty} = -\frac{1}{2} \left[X^T P X \right]_{t_0}^{\infty}
$$

\n
$$
= \frac{1}{2} \left[X_0^T P X_0 - X_{\infty}^T P X_{\infty} \right] = \frac{1}{2} \left(X_0^T P X_0 \right)
$$

LQR Design: Robustness of Closed Loop System

 \bullet **• Gain Margin:** ∞

• Phase Margin: $60^{\rm o}$

(Ref.: D. S. Naidu, Optimal Control Systems, CRC Press, 2003.)

Extensions of LQR Design

Dr. Radhakant Padhi Asst. Professor Dept. of Aerospace Engineering Indian Institute of Science - Bangalore

LQR Extensions:1. Cross Product Term in P.I.

$$
J=\frac{1}{2}\int_{t_0}^{\infty}\left(X^TQX+2X^TWU+U^TRU\right)dt
$$

Let us consider the expression:

$$
XT (Q-WR-1WT) X + (U + R-1WT X)T R (U + R-1WT X)
$$

= $XT QX + UT R U + (UT WT X + XT W U)$
= $XT QX + 2XT W U + UT R U$

LQR Extensions: 1. Cross Product Term in P.I.

$$
J = \frac{1}{2} \int_{t_0}^{\infty} \left[X^T \left(Q - W R^{-1} W^T \right) X + \left(U + R^{-1} W^T X \right)^T R \left(U + R^{-1} W^T X \right) \right] dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q_1 X + U_1 R U_1 \right) dt
$$

\n
$$
\dot{X} = AX + BU
$$

\n
$$
= AX + B \left(U_1 - R^{-1} W^T X \right)
$$

\n
$$
= (A - BR^{-1} W^T) X + BU_1
$$

\n
$$
= A_1 X + BU_1
$$

\n
$$
= -\left(K + R^{-1} W^T \right) X
$$

LQR Extensions: 2. Weightage on Rate of State

$$
J = \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q X + U^T R U + \dot{X}^T S \dot{X} \right) dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T Q X + U^T R U + (AX + BU)^T S (AX + BU) \right] dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left[\frac{X^T Q X + U^T R U + X^T A^T S A X + X^T A^T S B U}{+U^T B^T S A X + U^T B^T S B U} \right] dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left[X^T \left(Q + A^T S A \right) X + U^T \left(R + B^T S B \right) U + 2 X^T \left(A^T S B \right) U \right] dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left(X^T Q_1 X + U^T R_1 U + 2 X^T W U \right) dt
$$

\n
$$
\implies \text{Leads to a cross product case}
$$

$$
J = \frac{1}{2} \int_{0}^{\infty} \left(X^{T} Q X + U^{T} R U + U^{T} \hat{R} U \right) dt
$$

Let $\mathbf{X} = \left[\frac{X}{U} \right], \quad V = U$

$$
J = \frac{1}{2} \int_{0}^{\infty} \left(\mathbf{X}^{T} \begin{bmatrix} Q & 0 \\ 0 & R \end{bmatrix} \mathbf{X} + V^{T} \hat{R} V \right) dt
$$

$$
J = \frac{1}{2} \int_{0}^{\infty} \left(\mathbf{X}^{T} \hat{Q} \mathbf{X} + V^{T} \hat{R} V \right) dt
$$

 $\dot{X} = AX + BU, \quad X(0) = X_0$ ٠

 $U = V$ ٠

$$
\dot{\mathbf{X}} = \left[\begin{array}{cc} A & B \\ 0 & 0 \end{array}\right] \mathbf{X} + \left[\begin{array}{c} 0 \\ I \end{array}\right] V = \hat{A} \mathbf{X} + \hat{B} V
$$

Note :

(1) The dimension of the problem has increased **from** *n* **to** $(n+m)$

 (2) If $\{A, B\}$ is controllable, it can be shown that

the new system is also controllable.

Solution :

1 $V = \dot{U} = -\hat{R}^{-1}\hat{B}^T\hat{P}\mathbf{X}$

where \hat{P} is the solution of

1 $\hat{\mathbf{A}}^T\hat{P}+\hat{P}A-\hat{P}\hat{B}\hat{R}^{-1}\hat{B}^T\hat{P}+\hat{O}=0$ $\hat{A}^T \hat{P} + \hat{P}A - \hat{P} \hat{B} \hat{R}^{-1} \hat{B}^T \hat{P} + \hat{Q} =$

Hence

$$
\begin{aligned}\n\dot{U} &= -\hat{R}^{-1} \begin{bmatrix} 0 & I \end{bmatrix} \begin{bmatrix} \hat{P}_{11} & \hat{P}_{12} \\ \hat{P}_{12}^T & \hat{P}_{22} \end{bmatrix} \mathbf{X} = -\hat{R}^{-1} \begin{bmatrix} \hat{P}_{12}^T & \hat{P}_{22} \end{bmatrix} \begin{bmatrix} X \\ U \end{bmatrix} \\
&= -\hat{R}^{-1} \hat{P}_{12}^T X - \hat{R}^{-1} \hat{P}_{22} U\n\end{aligned}
$$

However, $\dot{U} = -\hat{R}^{-1} \hat{P}_{12}^T X - \hat{R}^{-1} \hat{P}_{22} U$ is a dynamic equation in U and hence is not easy for implementation. For this reason, we want an expression in the RHS only as a function of X and operations on it. $\hat{J} = -\hat{R}^{-1} \hat{P}_{12}^T X - \hat{R}^{-1}$

State equation: $X = AX + BU$ This suggests: $U = B^+\left(\dot{X} - AX\right)$ $= B^+$ $\left| X - \right|$ ٠

ė

(Note: This is only an approximate solution, unless $m \ge n$)

$$
\begin{split} \dot{U} &= -\hat{R}^{-1}\hat{P}_{12}^{T}X - \hat{R}^{-1}\hat{P}_{22}B^{+}\dot{X} - \hat{R}^{-1}\hat{P}_{22}B^{+}A\dot{X} \\ &= -\hat{R}^{-1}\left(\hat{P}_{12}^{T} + \hat{P}_{22}B^{+}A\right)X - \hat{R}^{-1}\hat{P}_{22}B^{+}\dot{X} \\ &= -K_{1}\dot{X} - K_{2}X \end{split}
$$

Integrating this expression both sides,

$$
U = -\underbrace{K_1 X}_{\text{Proportional}} - K_2 \underbrace{\int_{0}^{t} X(z) dz}_{\text{Integral}} + \underbrace{U_0}_{\text{Initial condition}}
$$

Note: U_0 can be obtained using a performance index ٠

Example 1 without the U term

LQR Extensions:4. Prescribed Degree of Stability

Condition: All the Eiganvalues of the closed loop system should lie to the left of line *AB*

$$
J = \frac{1}{2} \int_{t_0}^{\infty} e^{2\alpha t} \left[X^T Q X + U^T R U \right] dt \quad \text{where, } \alpha \ge 0
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left(\left[e^{\alpha t} X \right]^T Q \left[e^{\alpha t} X \right] + \left[e^{\alpha t} U \right]^T R \left[e^{\alpha t} U \right] \right) dt
$$

\n
$$
= \frac{1}{2} \int_{t_0}^{\infty} \left(\tilde{X}^T Q \tilde{X} + \tilde{U}^T R \tilde{U} \right) dt
$$

\nLet $\tilde{X} = e^{\alpha t} X$ Co-ordinate
\ntransformation
\ntransformation
\n $\tilde{U} = e^{\alpha t} U$

LQR Extensions: 4. Prescribed Degree of Stability

 $e^{at}\left(AX + BU\right) + \alpha e^{\alpha t}X$ $A\left(e^{\alpha t}X\right)+B\left(e^{\alpha t}U\right)+\alpha\left(e^{\alpha t}X\right)$ $\tilde{X} = (A + \alpha I) \tilde{X} + B \tilde{U}$ $X = e^{\alpha t} X + \alpha e^{\alpha t} X$ $= e^{\alpha t} (AX + BU) + \alpha e^{\alpha t}$ ۰ $\tilde{X} = e^{\alpha t} \vec{X}$ ٠ $\tilde{Y} = (A + \alpha I) \tilde{Y} + B I \tilde{I}$ $e^{\alpha t}U = -Ke^{\alpha t}X$ $U = -K\,X$ $U = -K X$ **Control Solution:** $\widetilde{I}=-\overline{K}\,\widetilde{\overline{Y}}$

LQR Extensions: 4. Prescribed Degree of Stability

 $\tilde{X} = \left[\left(A - BK \right) + \alpha I \right] \tilde{X}$ Modified System: $U = -K\,X$ $\lfloor (A-BK)+\alpha I \rfloor$ $\tilde{I} - K \tilde{Y}$ ٠ ~

 \tilde{X} \dot{X} = $(A - BK)X$ Actual S ystem: *U* = −*K X* ė $\Gamma = \Gamma + \Gamma - \Gamma$

 $[(A - BK) + \alpha I]$ will lie in the left-half plane. K is designed in such a way that eigenvalues of Hence, eigenvalues of $(A - BK)$ will lie to the left of a line parallel to the imaginary axis, which is located away by distance α from the imaginary axis.

LQR Design for Command Tracking

Dr. Radhakant Padhi Asst. Professor Dept. of Aerospace Engineering Indian Institute of Science - Bangalore

LQR Design for Command Tracking

Problem:

To design U such that a part of the state vector of the

 $\frac{X - AX + BU}{\text{tracks a commanded reference signal}}$. ó

i.e.
$$
X_T \to r_c
$$
, where $X = \left[\frac{X_T}{X_N}\right]$

Solution:

1) Formulate a standard LQR problem. **However, select the Q matrix properly.**

Typically
$$
Q = \begin{bmatrix} Q_{TT} & 0 \\ 0 & 0 \end{bmatrix}
$$

2) Implement the controller as
$$
U = -K \begin{bmatrix} X_T - r_c \\ X_N \end{bmatrix}
$$

⎤

⎦

LQR Design for Command Tracking

LQR Design for Command Tracking with Integral Feedback

Solution (with integral controller):

1) Augment the system dynamics with integral states

$$
\begin{bmatrix} \dot{X}_T \\ \dot{X}_N \\ \dot{X}_I \end{bmatrix} = \begin{bmatrix} A_{TT} & A_{TN} & 0 \\ A_{NT} & A_{NN} & 0 \\ I & 0 & 0 \end{bmatrix} \begin{bmatrix} X_T \\ X_N \\ X_I \end{bmatrix} + \begin{bmatrix} B_T \\ B_N \\ 0 \end{bmatrix} U
$$

2) Select the Q matrix properly *Simulterially* X_T and X_t states)

3) Control solution
$$
U = -K \left[\left(X_T - r_c \right)^T \quad X_N^T \quad \left(\int_0^t \left(X_T - r_c \right) dt \right)^T \right]^T
$$

Dr. Radhakant Padhi Asst. Professor Dept. of Aerospace Engineering Indian Institute of Science - Bangalore

Reference

Optimum Intercept Laws for Accelerating Targets

VITALIJ GARBER*

AIAA JOURNAL VOL. 6, NO. 11 NOVEMBER 1968

• To derive the state X of a linear (rather $\int \textbf{R} \cdot \textbf{R} \cdot \textbf{R} \cdot \textbf{R} \cdot \textbf{S} = A X + B U + C$ to the origin by minimizing the following quadratic performance index (cost function)

$$
J = \frac{1}{2} \Big(X_f^T S_f X_f \Big) + \frac{1}{2} \int_{t_0}^{t_f} \Big(X^T Q X + U^T R U \Big) dt
$$

where

 S_f , $Q \ge 0$ (psdf), $R > 0$ (pdf)

- **Performance Index (to minimize):** \bullet Path Constraint: $\dot{X} = AX + BU + C$ $(X_f^T S_f X_f) + \frac{1}{2} \int (X^T Q X + U^T R U)$ 0 $1, 1, 1, 1, 1$ 2^{2} 2 t ^{f} $T \cap V$, $\frac{1}{I} \cap V$ $\cap V$, I $J = \frac{1}{2} (X_f^T S_f X_f) + \frac{1}{2} \int (X^T Q X + U^T R U) dt$ *t*
- \bullet \bullet Boundary Conditions: $X(0) = X_0$: Specified t_f : Fixed, $X(t_f)$: Free

- Terminal penalty: $\varphi(X_f) = \frac{1}{2}(X_f^T S_f X_f)$ 1 2 $\varphi(X_f) = \frac{1}{2} \left(X_f^T S_f X_f \right)$
- Hamiltonian: $H = \frac{1}{2}(X^TQX + U^TRU) + \lambda^T(AX + BU + C)$ 1 2 $H = -\left(X^{T}Q X + U^{T} R U\right) + \lambda^{T}\left(AX + BU + C\right)$
- State Equation: $\dot{X} = AX + BU + C$
- Costate Equation: $\lambda = -(\partial H / \partial X) = -(QX + A^T)$) ٠ $\partial_t = -(\partial H / \partial X) = -(\partial X + A' \lambda)$
- Optimal Control Eq.: $(\partial H / \partial U) = 0 \Rightarrow U = -R^{-1}B^{T}\lambda$ $\partial H / \partial U$ = 0 $\Rightarrow U = -R^{-}$
- **Boundary Condition:** $\lambda_f = (\partial \varphi / \partial X_f)$ $\lambda_f = \left(\frac{\partial \varphi}{\partial X_f}\right) = S_f X_f$

LQR Design for Inhomogeneous Systems **Guess** $\lambda(t) = P(t)X(t) + K(t)$ $\dot{P} = \dot{P}X + P\left(AX + BU + C\right) + \dot{K}$ $\dot{P}X + P\left(AX - BR^{-1}B^{T}\lambda\right) + PC + \dot{K}$ $-(QX + A^T(PX + K)) = \dot{P}X + P(AX - BR^{-1}B^T(PX + K)) + PC + \dot{K}$ $\lambda = PX + PX +K$ $=PX + P(AX - BR^{-1}B' \lambda) + PC +$ $\dot{A} - \dot{P}Y + P\dot{Y} + \dot{K}$ $\dot{D}Y + P(AY + RII + C) + \dot{K}$ $\dot{D}V + P(\Delta V - RR^{-1}R^{T}\lambda) + PC + \dot{K}$ $\dot{D}V + P(\Lambda V - RR^{-1}R^{T}(DY + K)) + PC + \dot{K}$ $\left(\dot{P} + PA + A^TP - PBR^{-1}B^TP + Q\right)X$ $(\dot{K} + A^T K - PBR^{-1}B^T P + PC) = 0$ $\dot{K} + A^T K - PBR^{-1}B^T P + PC$ $+ A' K - PBR^{-1}B' P + PC$ = ٠ ٠

 \bullet Riccati equation

$$
\dot{P} + PA + A^T P - PBR^{-1}B^T P + Q = 0
$$

• Auxiliary equation

$$
\overline{\dot{K} + \left(A^{T} - PBR^{-1}B^{T}\right)K + PC} = 0
$$

zBoundary conditions

$$
P(t_f)X_f + K(t_f) = S_f X_f \t(X_f \text{ is free})
$$

$$
P(t_f) = S_f \t K(t_f) = 0
$$

Control Solution:

$$
U = -R^{-1}B^{T} \lambda
$$

= -R^{-1}B^{T} (PX + K)
= -R^{-1}B^{T}PX - R^{-1}B^{T}K

Note: There is a residual controller even after $X \rightarrow 0$. This part of the controller offsets the continuous disturbance.

