



Jet Aircraft Propulsion

Prof. Bhaskar Roy, Prof. A M Pradeep

Department of Aerospace Engineering,
IIT Bombay

Lect-27

In this lecture...

- Intakes for powerplant
 - Transport aircraft
 - Military aircraft

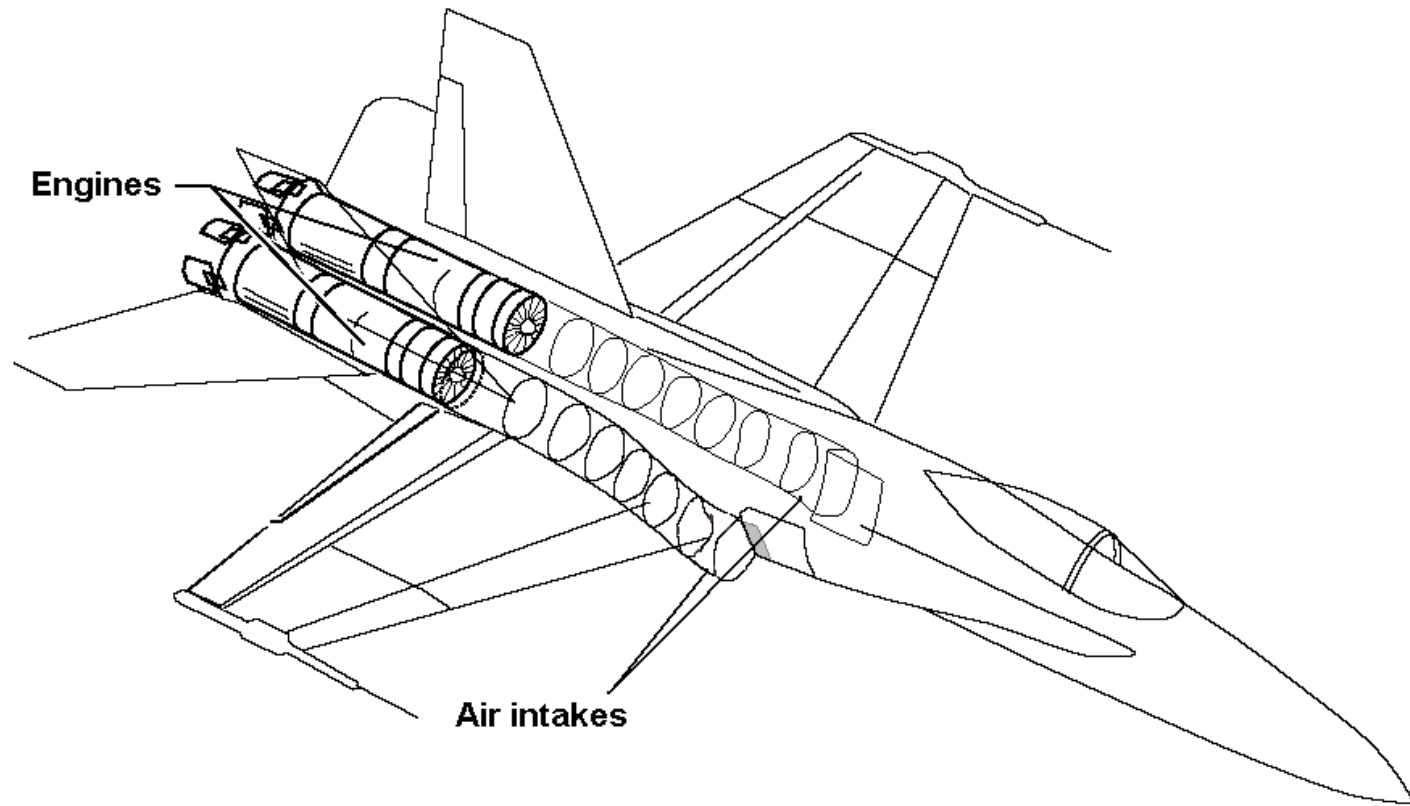
Intakes

- Air intakes form the first component of all air breathing propulsion systems.
- The word Intake is normally used in the UK and Inlet in the United States.
- Air intakes are usually manufactured by the airframe manufacturer in coordination with the engine manufacturer.
- An aircraft may have one or more intakes depending upon the engine.

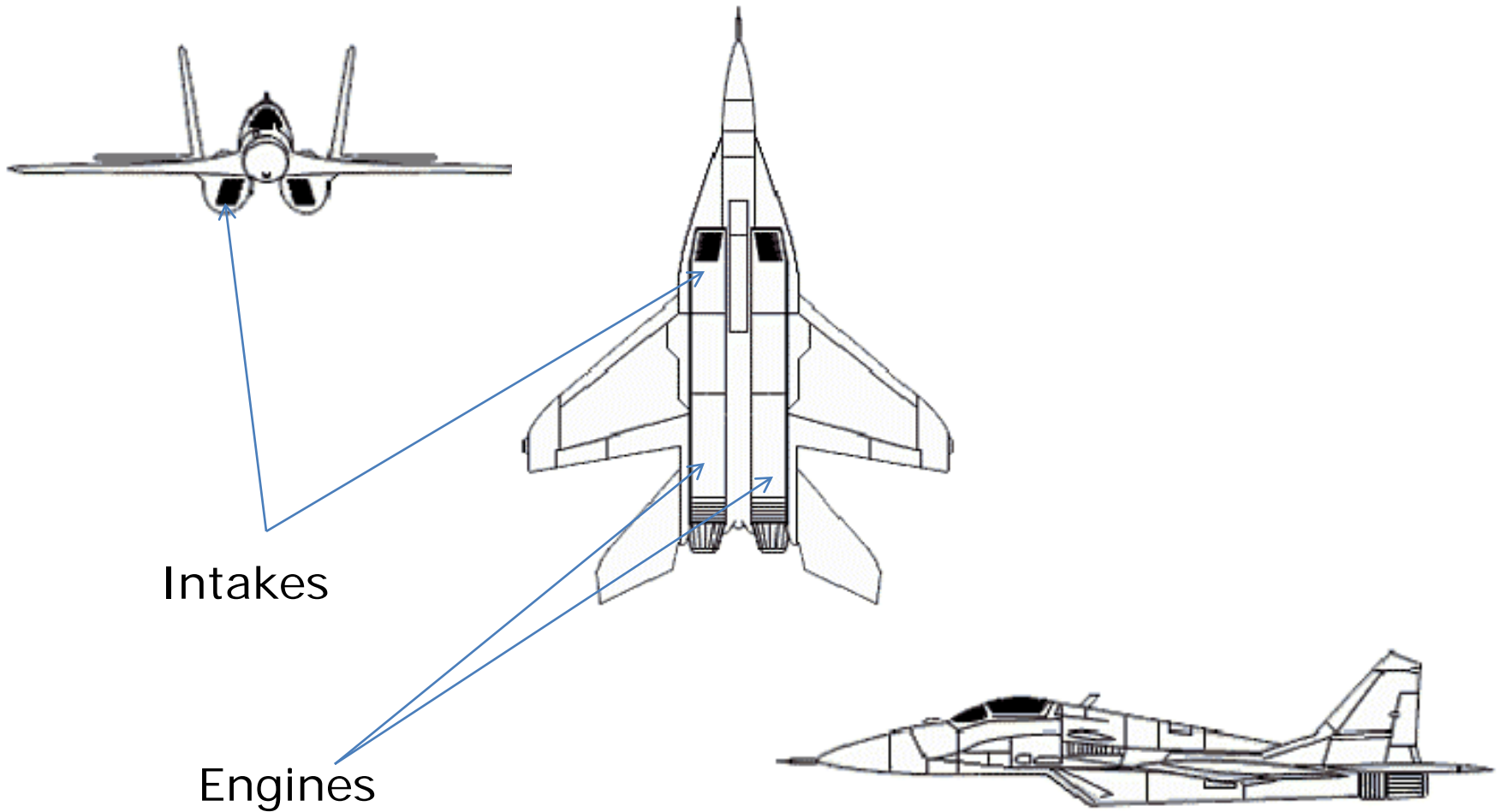
Intakes

- Air intakes are required to capture freestream air, sometimes change its direction and then supply this to the engine.
- This must be with as little flow distortion (non-uniformity) as possible.
- The intake must also not result in excessive external drag to the aircraft.
- Intake must ensure proper operation over the entire flight regime.
- Modern aircraft intakes also contain noise-absorbing materials.

Intakes



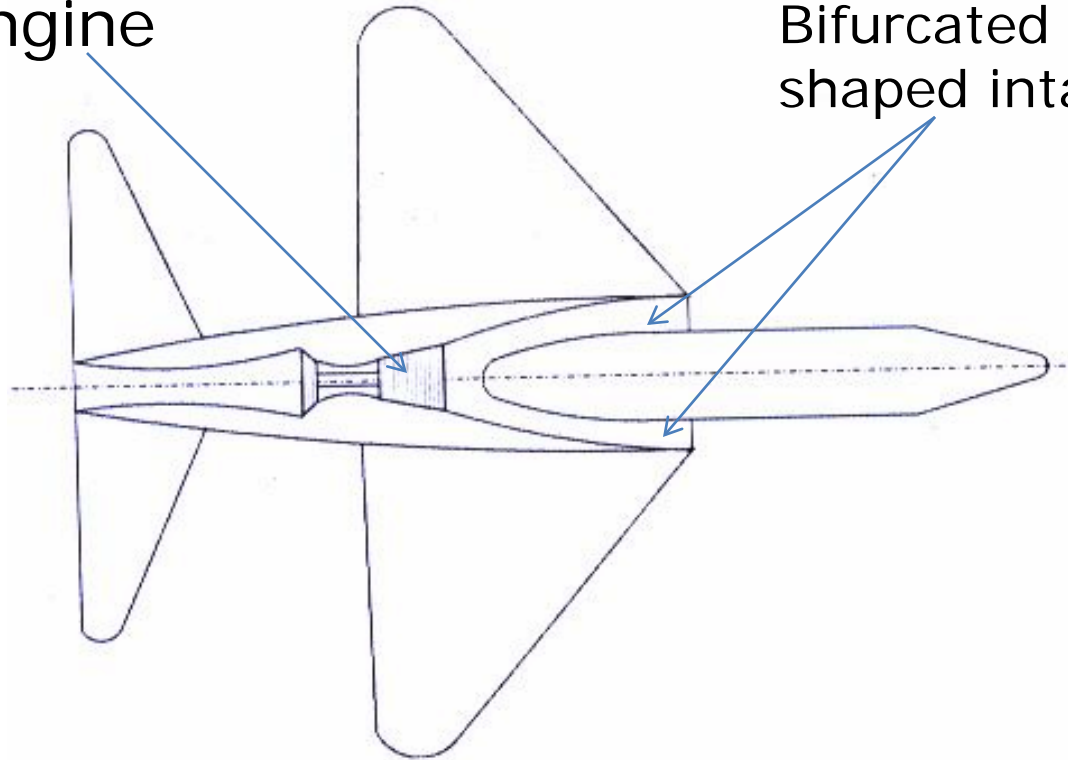
Intakes



Intakes

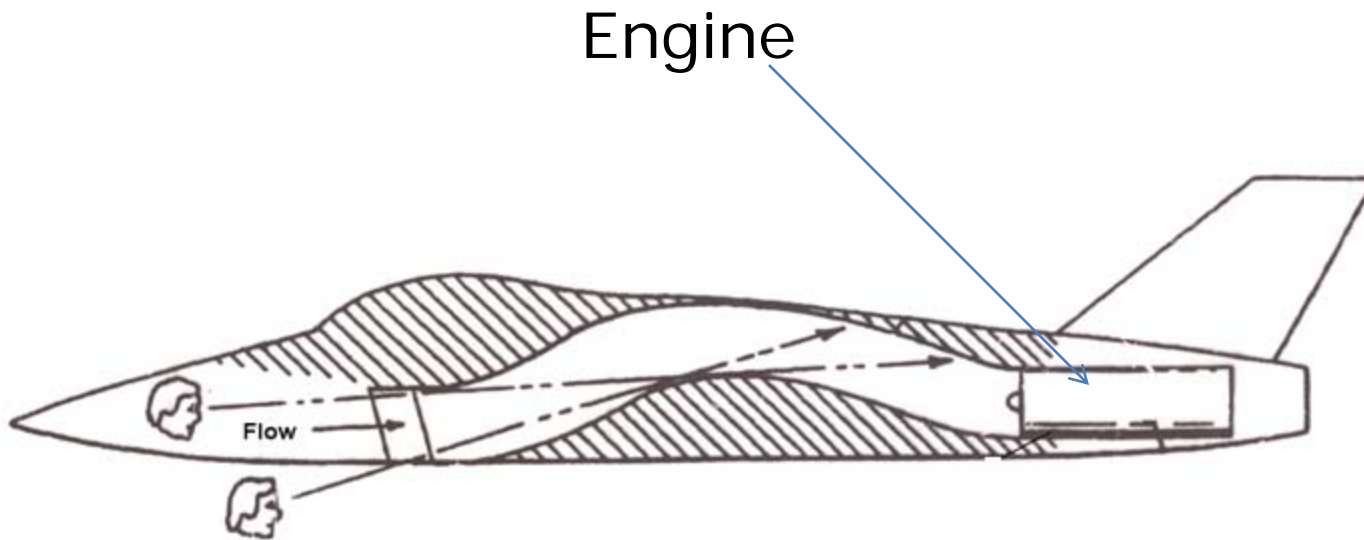
Single engine

Bifurcated intakes or Y shaped intakes



Bifurcated or Y-shaped intakes

Intakes



Serpentine intakes for next generation military aircraft

Intakes

- Intakes used in transport aircraft are quite different from the military air intakes.
- All operational transport aircraft are subsonic.
- Subsonic intakes consist of surfaces with smooth continuous curves.
- Usually such intakes have a thick leading edge: lip.
- Intakes of turboprops are slightly more complicated due to the presence of the propeller and the gearbox.

Intakes



Transport aircraft intakes

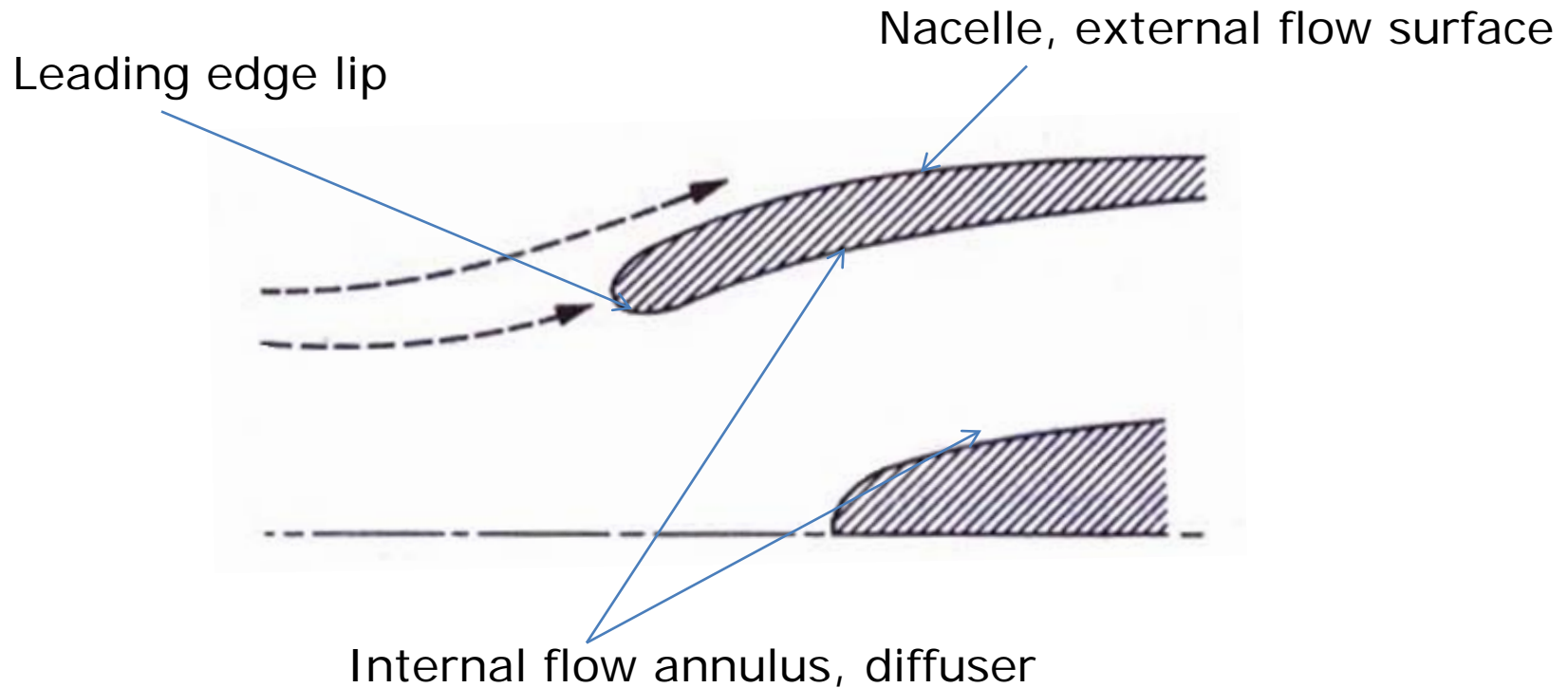


Military aircraft intakes

Subsonic intakes

- Most subsonic intakes have fixed geometries.
- Some of the high bypass ratio turbofans have blow-in-doors.
- These are designed to deliver additional air to the engine during take-off and climb.
- This is to cater to the fact that under these conditions, the engine requires maximum thrust and hence more mass flow rate of air.

Subsonic intakes



Typical subsonic (Pitot type) intake geometry

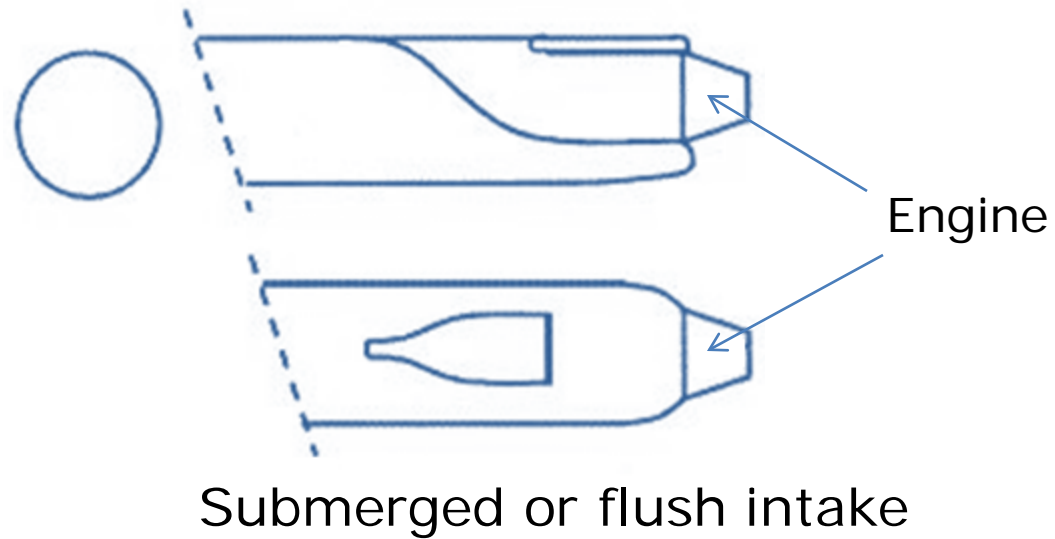
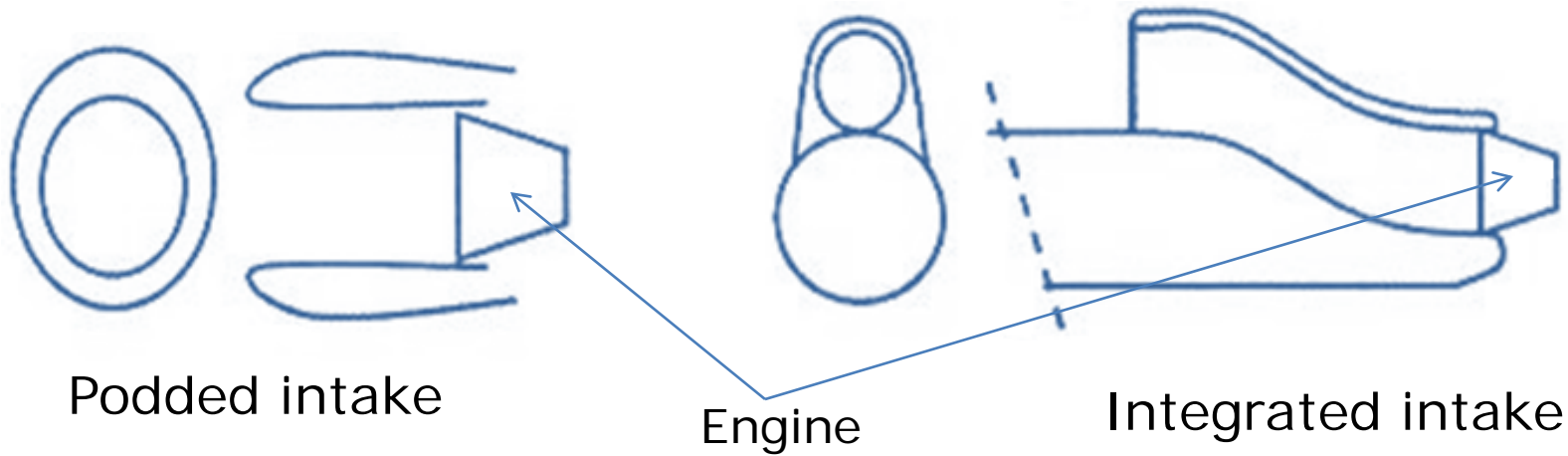
Pitot intakes

- The most common type of a subsonic intake is the Pitot intake.
- Pitot intakes make the best use of the ram effect due to forward motion.
- These intakes also suffers the minimum loss of ram pressure during changes in altitude.
- However these intakes are primarily for subsonic operation.

Pitot intakes

- There are three types of Pitot intakes:
 - Podded intakes
 - Integrated intakes
 - Flush intakes
- Podded intakes are commonly used in transport aircraft (civil or military).
- Integrated intakes are used in combat aircraft.
- Flush intakes are used in missiles.

Pitot intakes



Podded intakes

- Usually the friction losses in podded intakes are insignificant.
- Flow separation may drastically affect the performance.
- The leading edge of the intake captures a certain streamtube and then divides this stream into internal flow and external flow.
- The design of the duct must be such that it preserves good aerodynamics of the airframe and an internal flow with minimal loss.

Integrated and flush intakes

- For integrated intakes, the internal flow problems are of concern.
 - The duct is usually longer and with bend(s).
 - The aircraft fuselage ahead of the intake feeds a boundary layer into the intake.
- Curvature of the intake also leads to generation of secondary flows leading to flow distortion.
- Flush intakes are used in missiles as these can be easily accommodated into missile airframes as well as for canister launching.

Supersonic intakes

- Supersonic intakes are usually more complicated than subsonic intakes.
- Design of supersonic intakes often involve trade-offs between efficiency, complexity, weight and cost.
- Supersonic intakes consists to two segments: a supersonic diffuser where the flow is decelerated from supersonic to subsonic through a series of shocks; this is followed by a subsonic diffuser where the flow is decelerated from high subsonic to lower subsonic speeds.

Supersonic intakes

- Supersonic intake design is more complicated due to the following reasons:
 - Shock waves cause significant loss in total pressure.
 - Large variation in the capture area between subsonic and supersonic flight.
 - With higher Mach number the inlet compression ratio is a larger fraction of overall pressure ratio, thrust becomes more sensitive to diffuser performance.
 - Efficient operation of the intake in both subsonic and supersonic flight regimes.

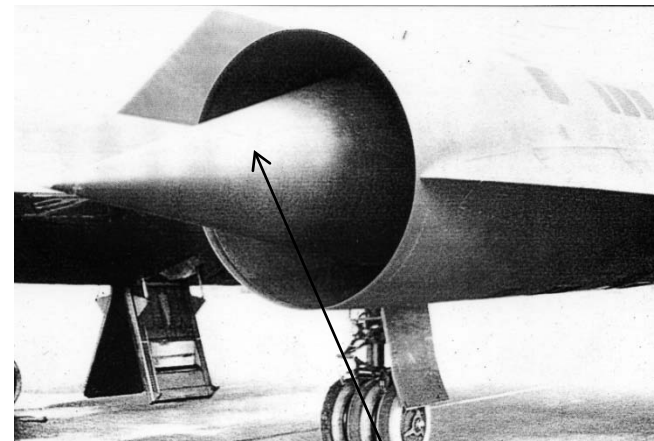
Supersonic intakes

- Supersonic intakes may be classified in the following ways:
 - Axisymmetric or two-dimensional
 - Axisymmetric: central cone for shock fixture
 - Two-dimensional: rectangular cross-section
 - Variable or fixed geometry
 - Variable: the central cone may be movable or in a rectangular intake, one of the walls may be adjustable
 - Fixed: Geometry is fixed

Supersonic intakes

- Internal, external or mixed compression
 - Depending upon the location of the shocks
 - Internal compression intakes have shocks that are located within the intake geometry
 - External compression intakes have shocks located outside the intake
 - Mixed compression intakes have shock that are located within as well as outside the intake geometry.

Supersonic intakes



Centerbody

Intakes

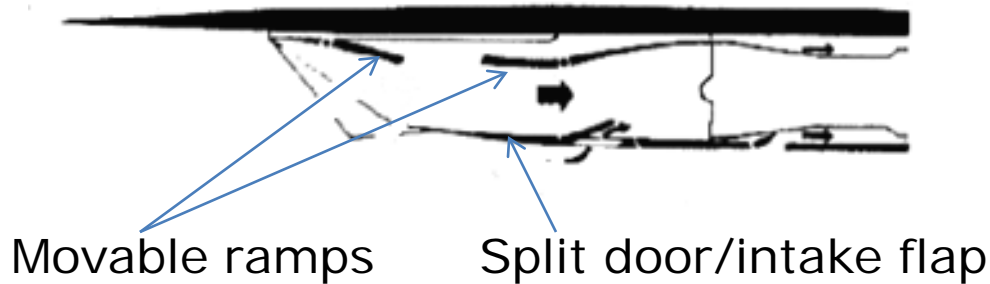
Axisymmetric intake with spiked centerbody

Supersonic intakes

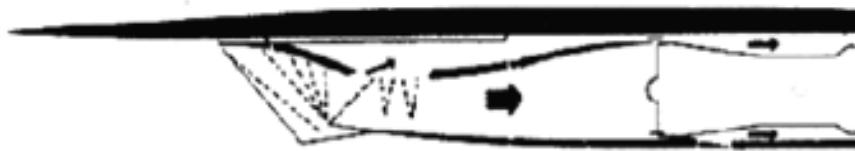


2D intakes of Concorde

Supersonic intakes



Intake during take-off



Supersonic cruise



Engine shut down

Modes of operation of a variable geometry 2D intake

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In the next lecture...

- Performance of intakes
 - Performance parameters
 - Sources of losses
 - Starting problem in supersonic intakes