



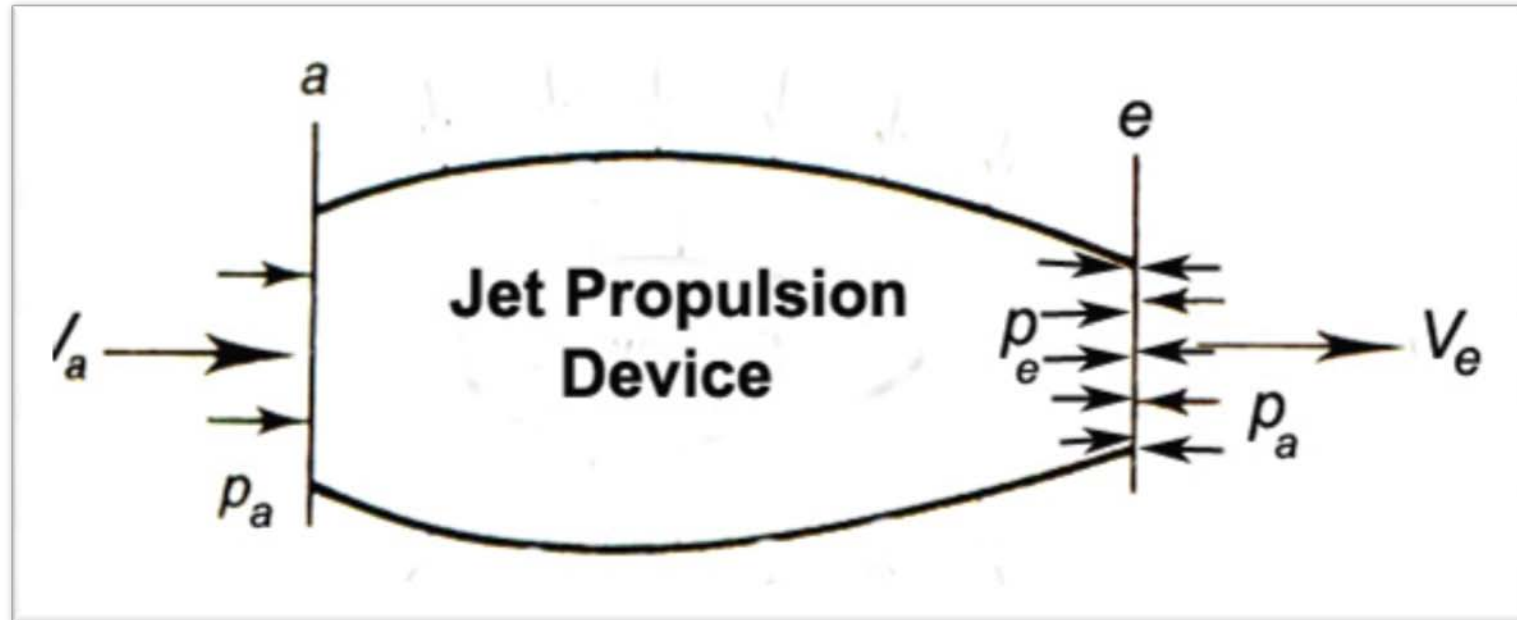
# Introduction to Aerospace Propulsion

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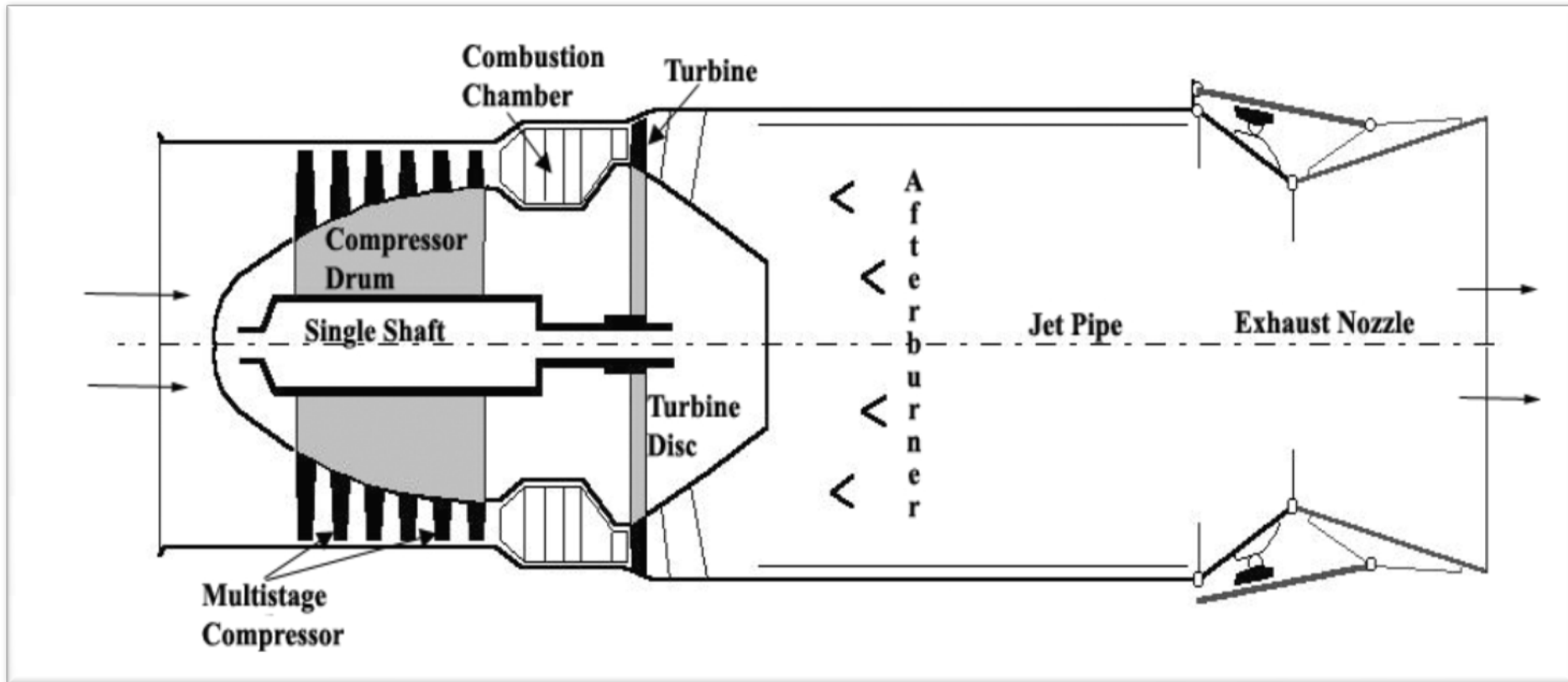
Lecture No - 3



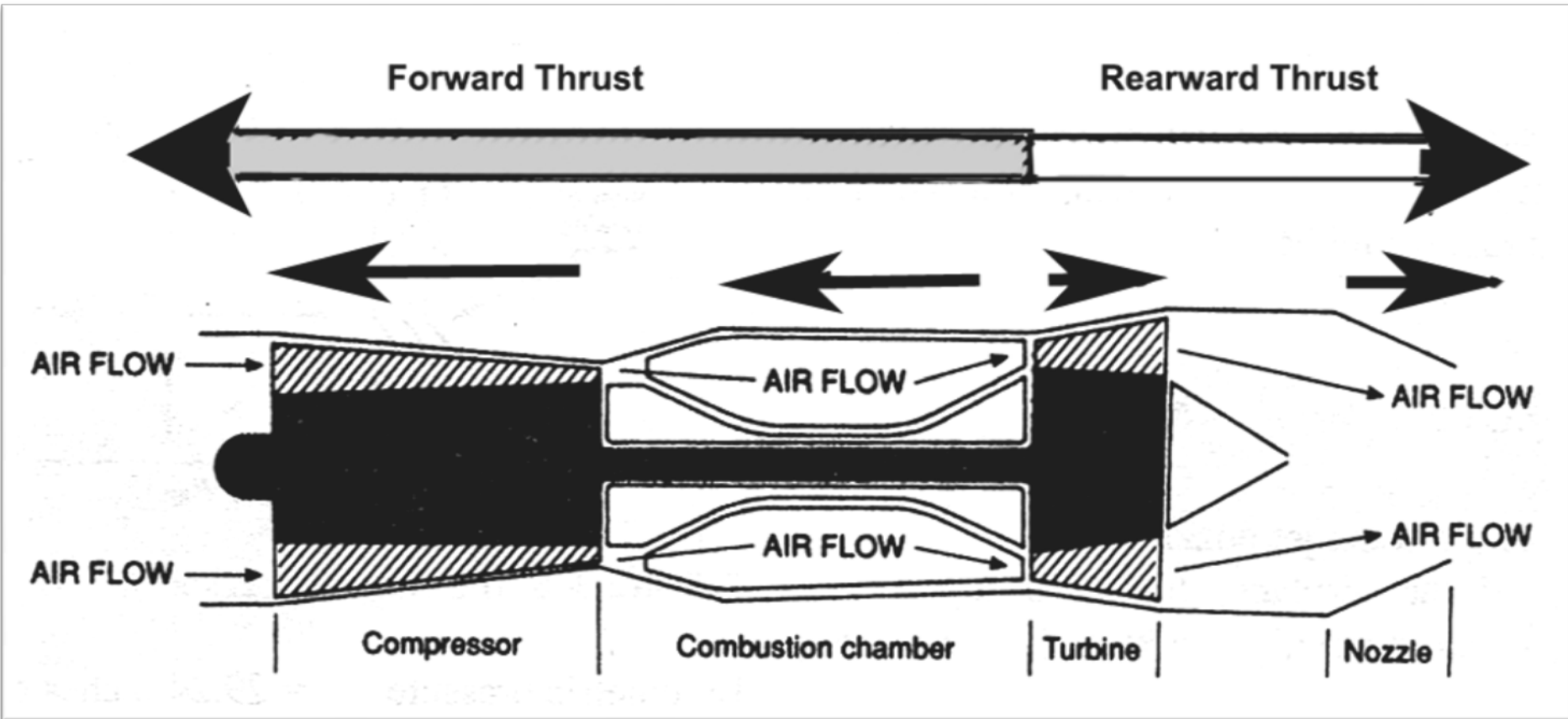


$$\text{Thrust} = \dot{m}(V_e - V_a)$$

## Simple Gas Turbine based Turbojet Engine



## Mechanism of Creation of Thrust



The net thrust  $F$  due to change in momentum is

$$F = \dot{m} V_e - V_a \dot{m} + A_e (P_e - P_a)$$

Intake Ram drag
↓

↑
↑

Gross Momentum Thrust
Pressure Thrust

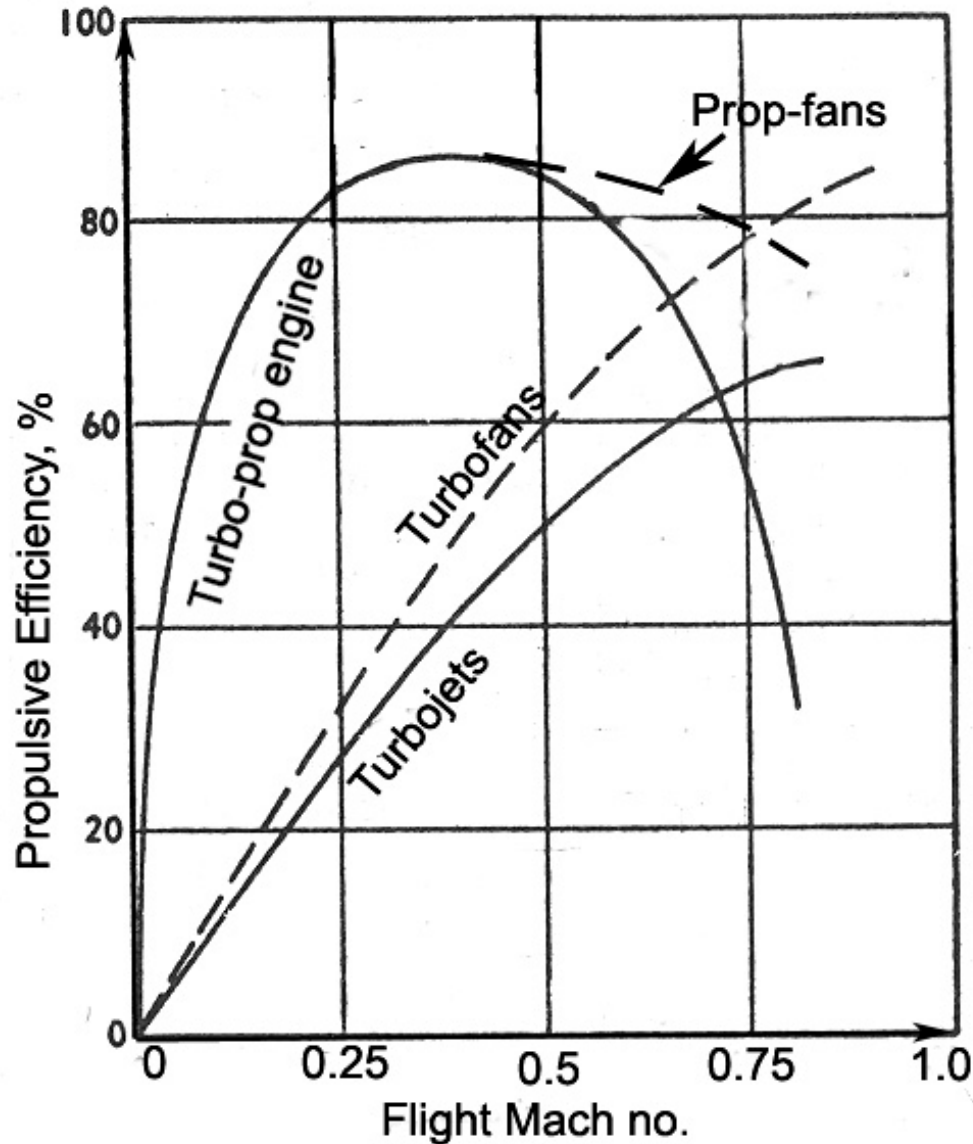
The propulsive efficiency  $\eta_p$  can be defined as the ratio of the useful propulsive energy or thrust power ( $F \cdot V_a$ ) to the sum of that energy and the unused kinetic energy of the jet,  $\frac{\dot{m} (V_e - V_a)^2}{2}$ . The latter is the kinetic energy relative to earth.

## propulsive efficiency

$$\eta_p = \frac{\dot{m} \cdot V_a \cdot (V_e - V_a)}{\dot{m} \left( V_a \cdot (V_e - V_a) + \frac{(V_e - V_a)^2}{2} \right)} = \frac{2}{1 + \left( \frac{V_e}{V_a} \right)}$$

When  $V_e \gg V_a$  i.e. a very large acceleration and, so with even with low mass flow, Thrust produced,  $F =$  very high- , but propulsive efficiency,  $\eta_p =$  low , typical jet engine, which produces compact thrusters

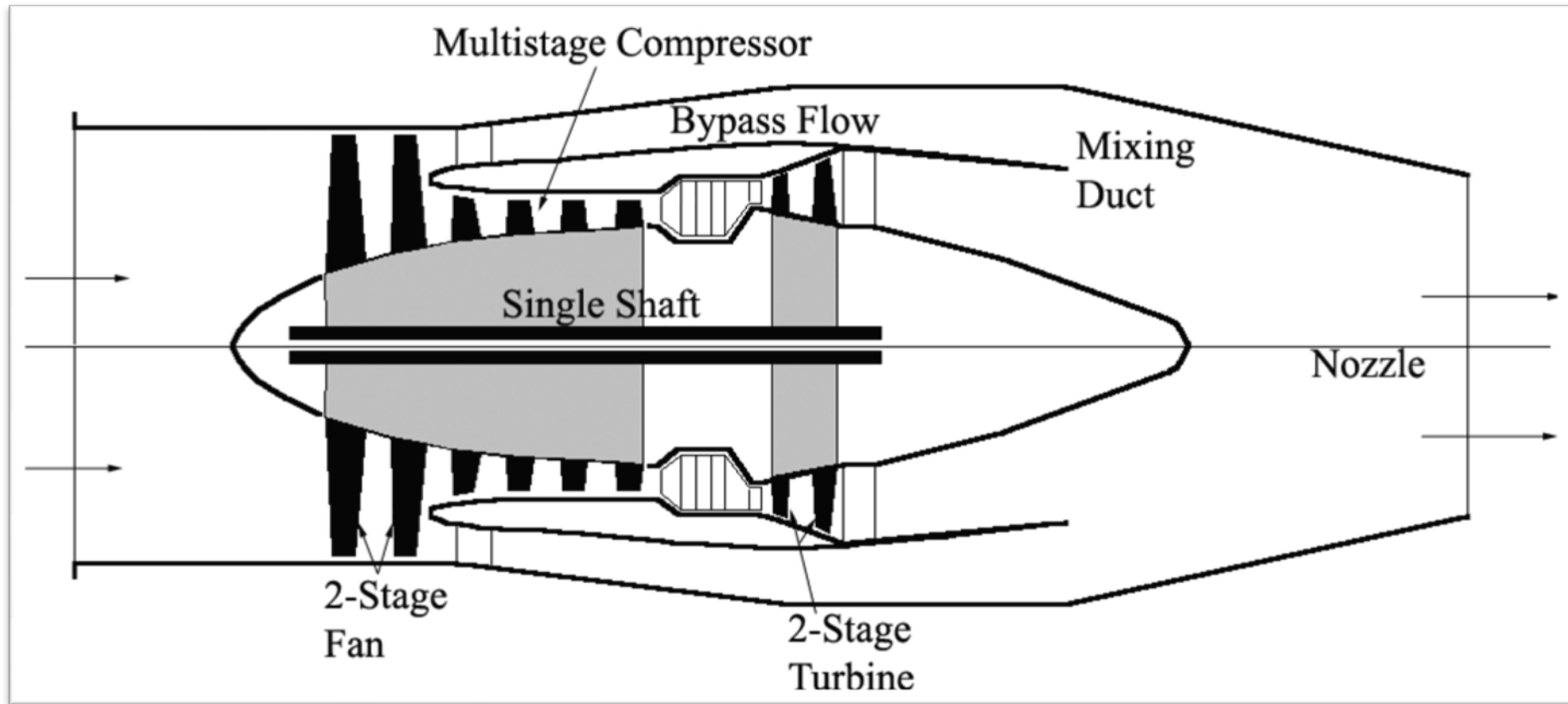
When,  $V_e = V_a$  the propulsive efficiency is 100%, but Thrust,  $F \approx 0$ ; - has given rise to turbofans, where large part of the thrust is produced with high mass flow, low air acceleration and high propulsive efficiency, and only a small part of thrust is produced with high jet effect.



The graph captures the utility zones of various kinds of propulsive devices.

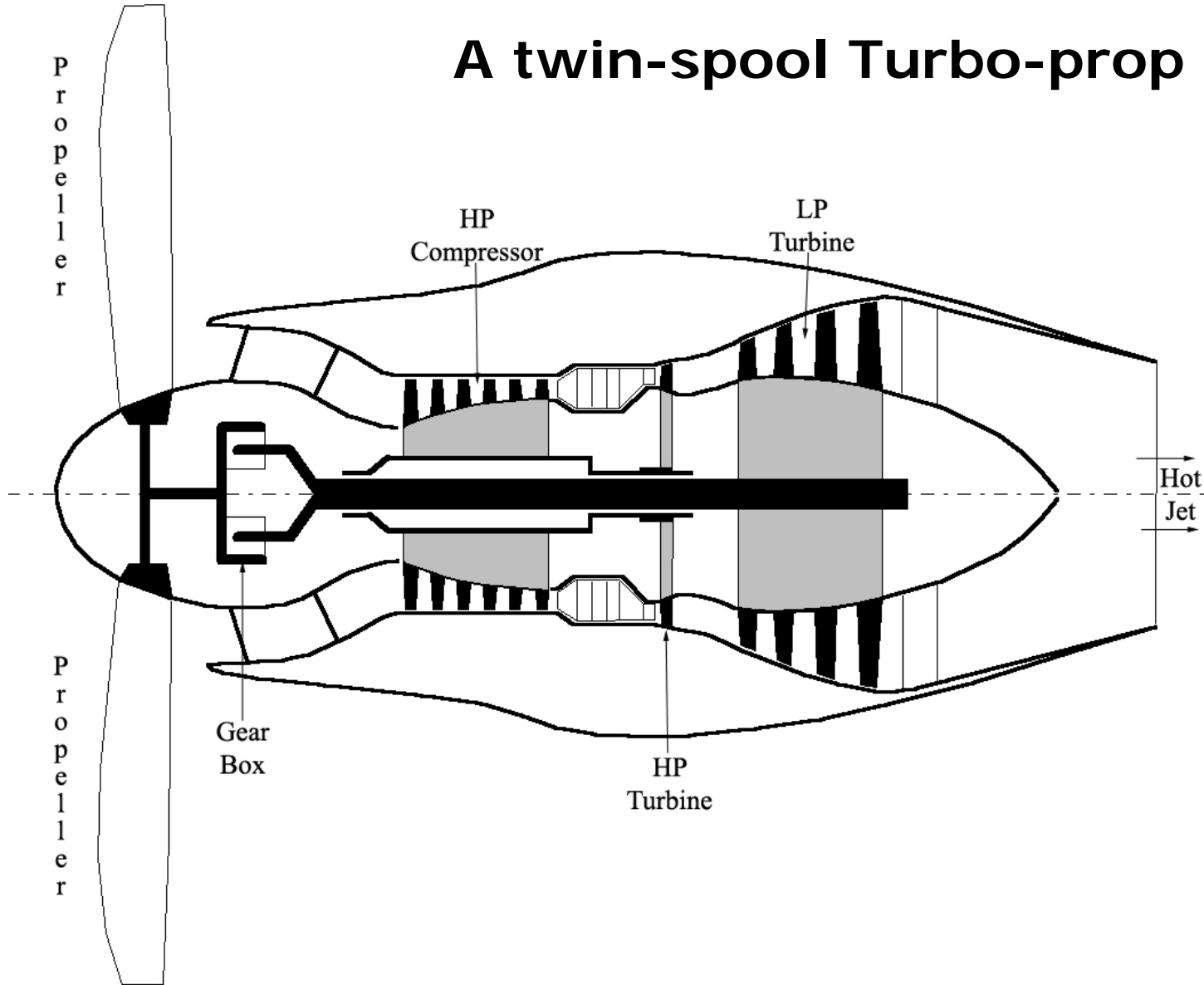
- At low subsonic speeds propellers are the better thrusting devices.
- At medium speeds (high subsonic) turbofans are the better thrusters
- At high (supersonic) speeds, there is no alternative but to go for pure jet engines

## A single spool bypass turbojet engine

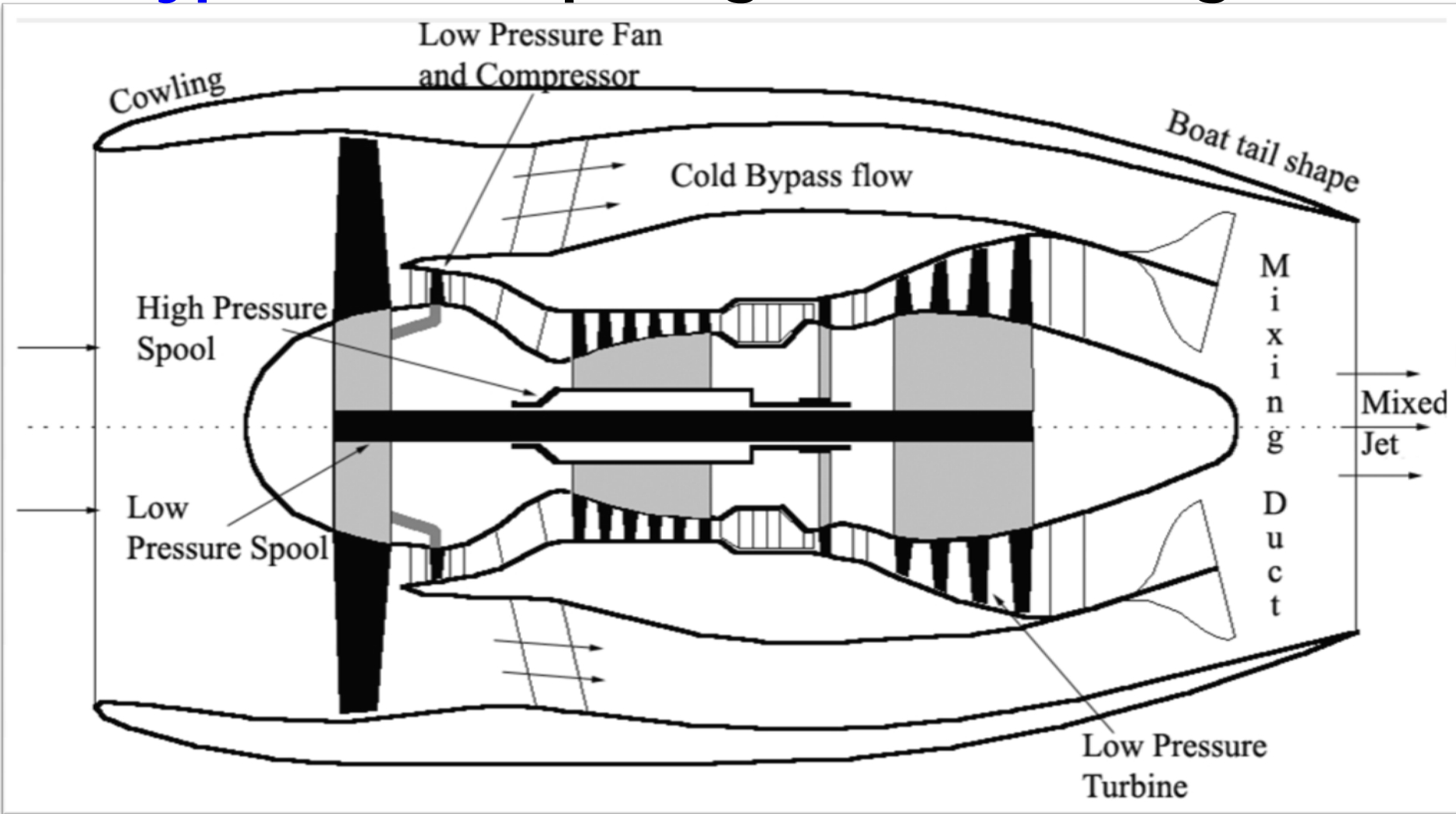




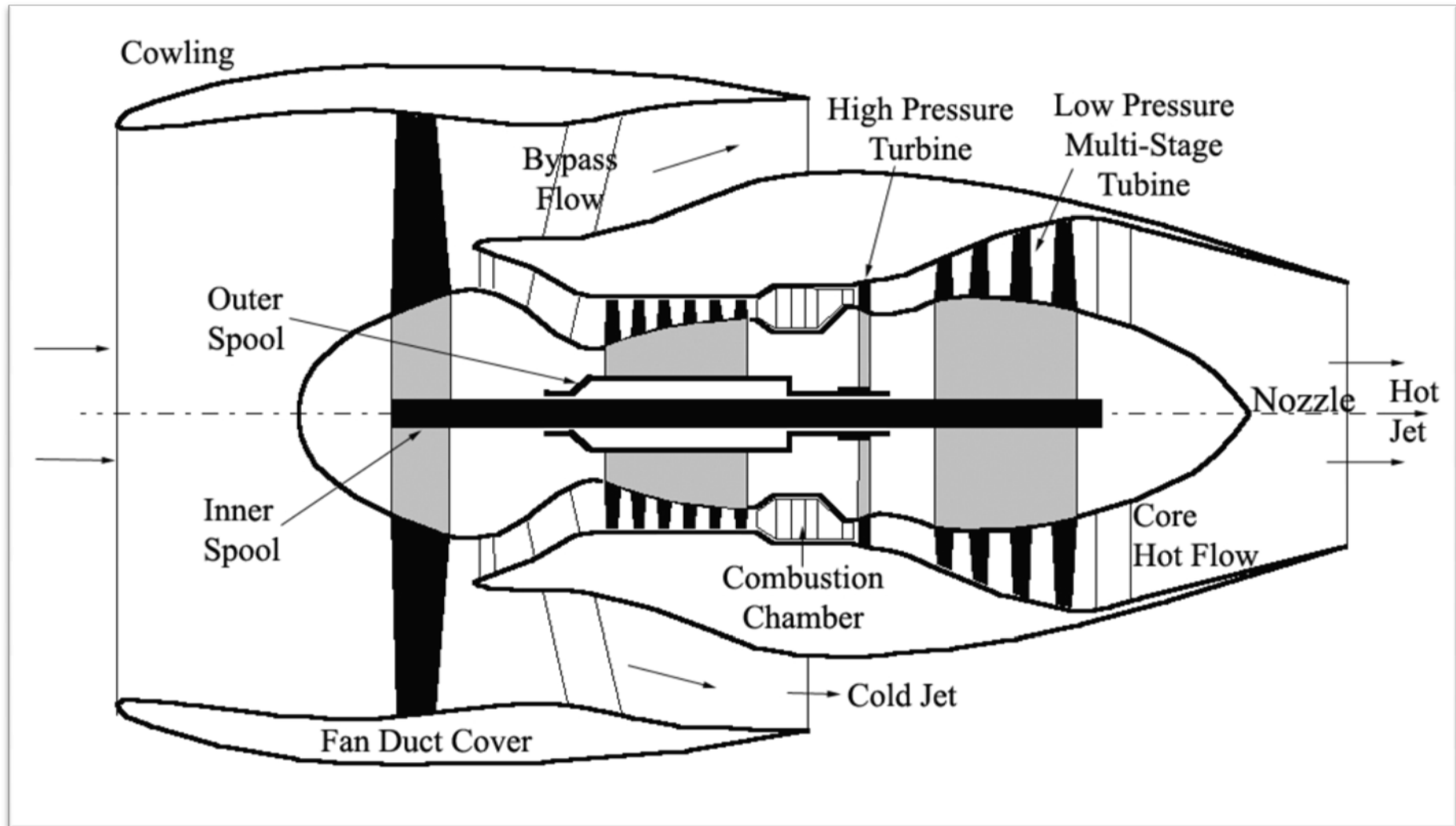
## A twin-spool Turbo-prop engine



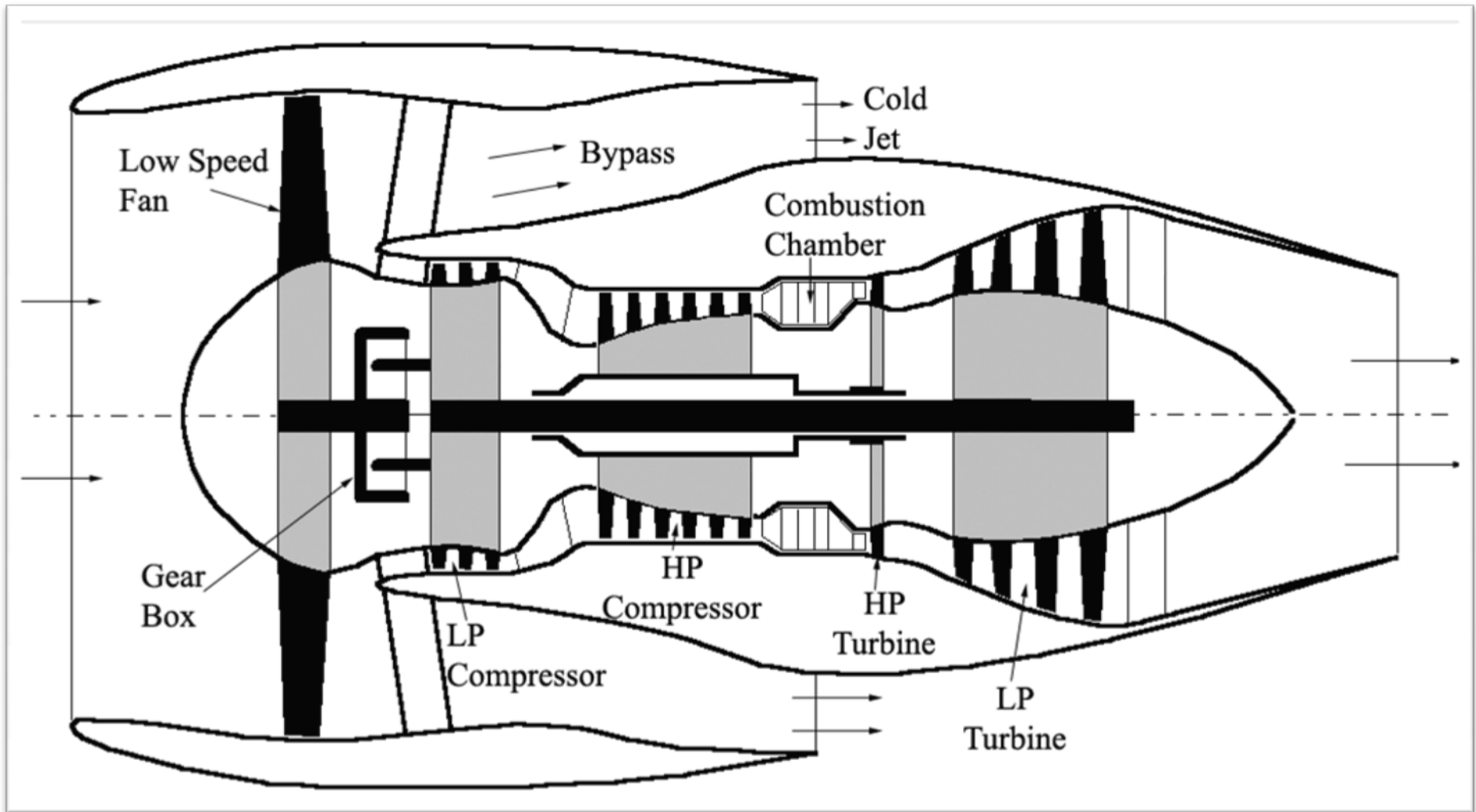
## A **bypass** twin spool gas turbine engine



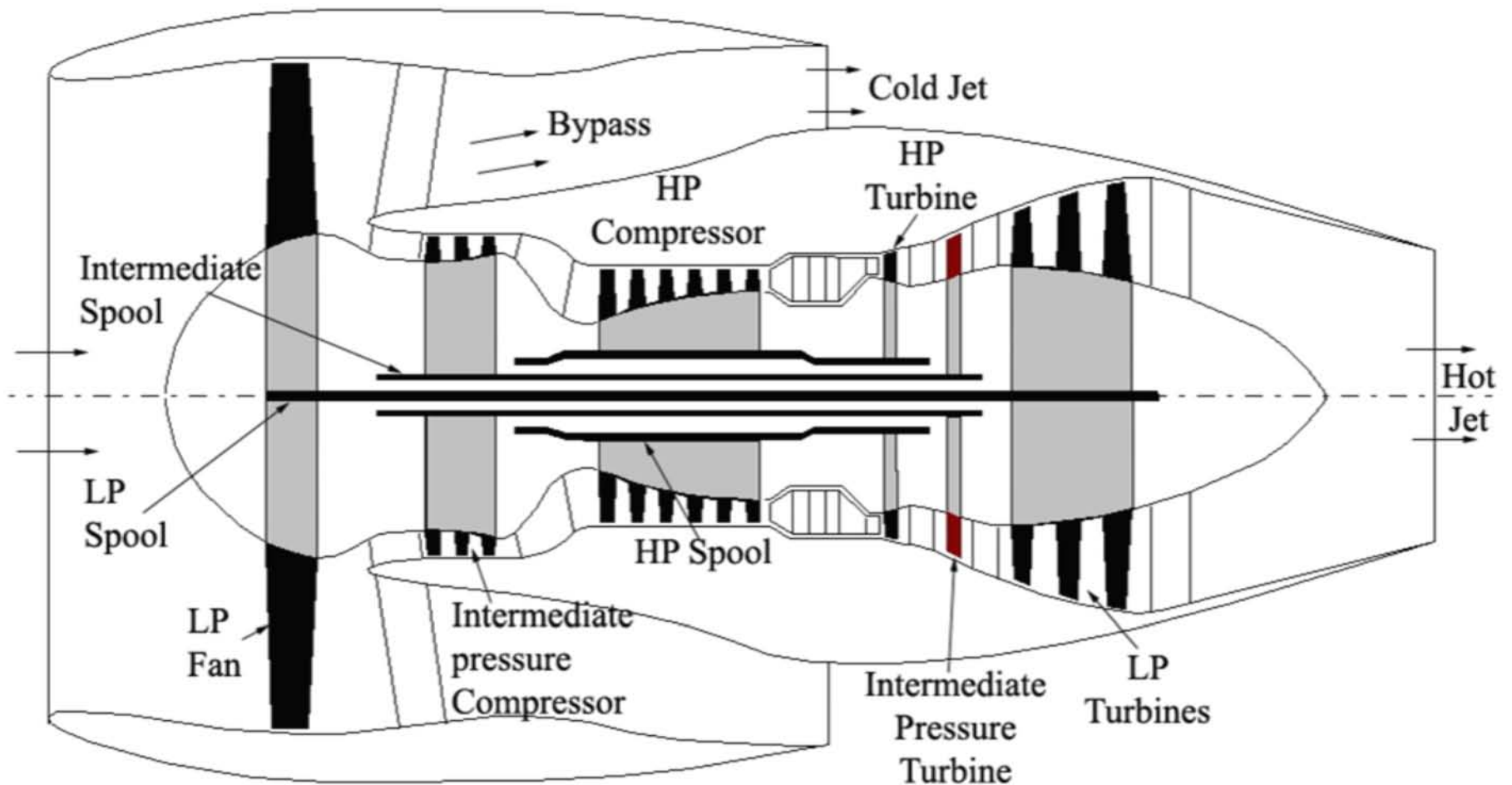
## A high bypass twin spool Turbofan engine



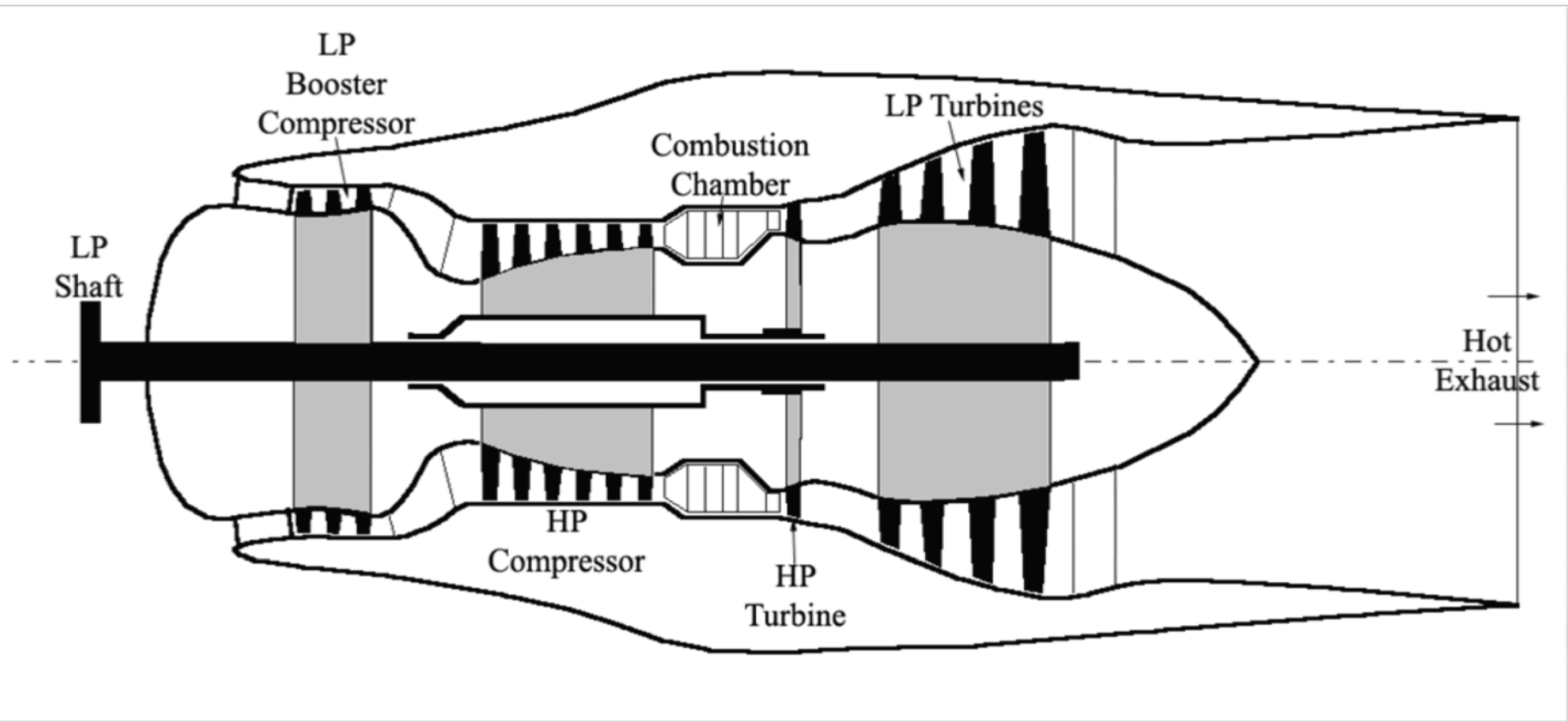
## An Ultra-high bypass twin spool geared turbofan



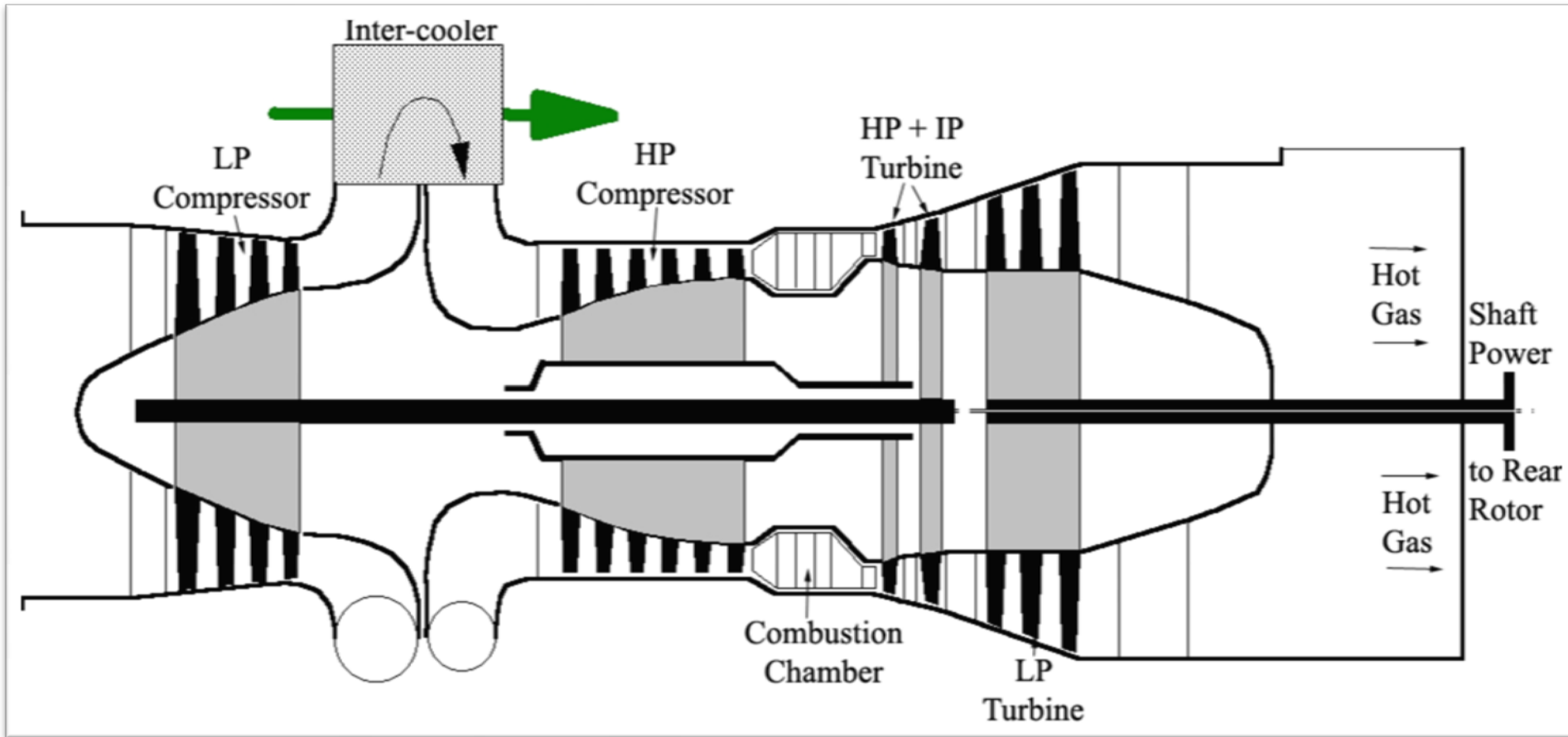
## A three-spool turbofan bypass engine



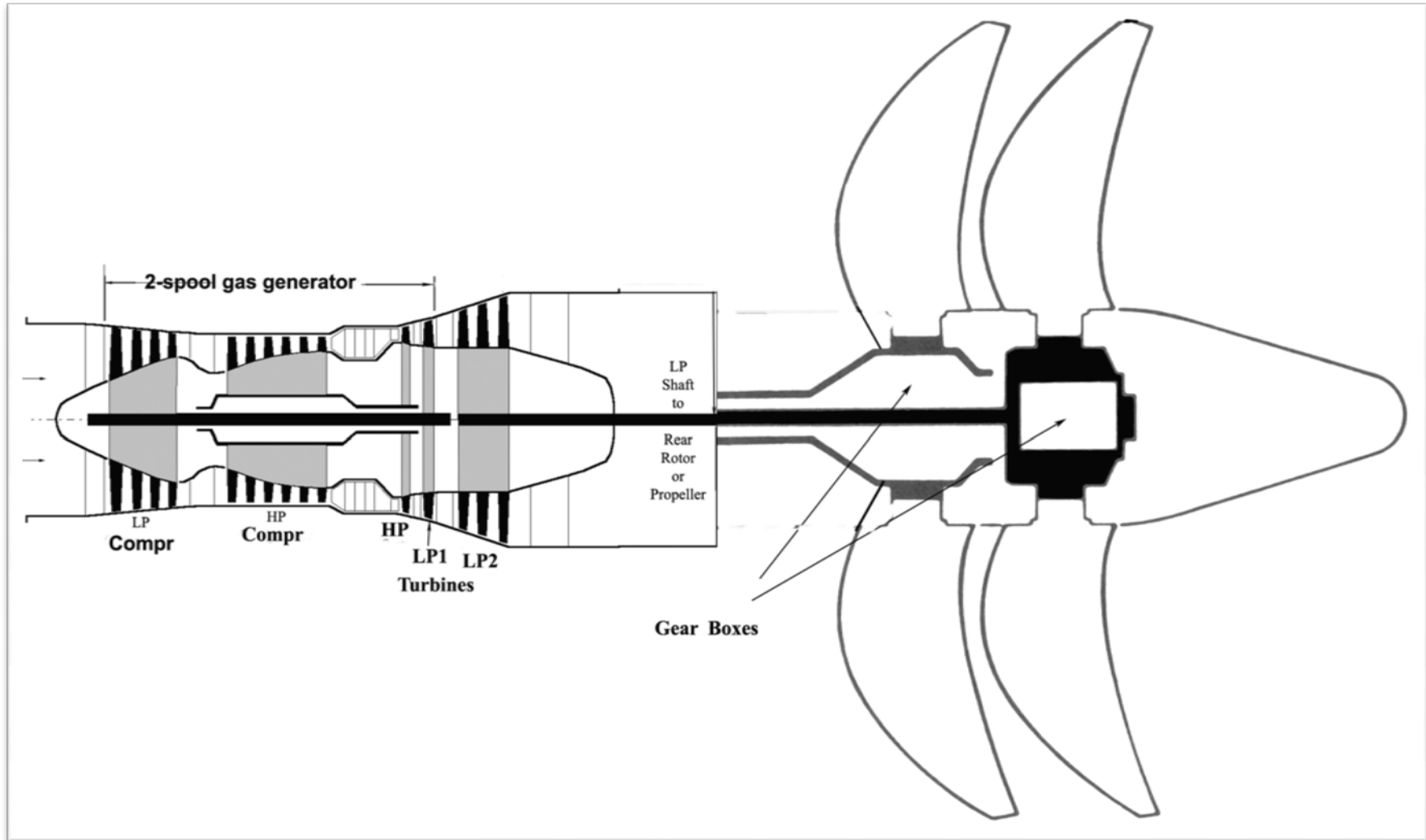
## Two spool turboshaft engine (for propeller)



## Twin spool powerplant with inter-cooling

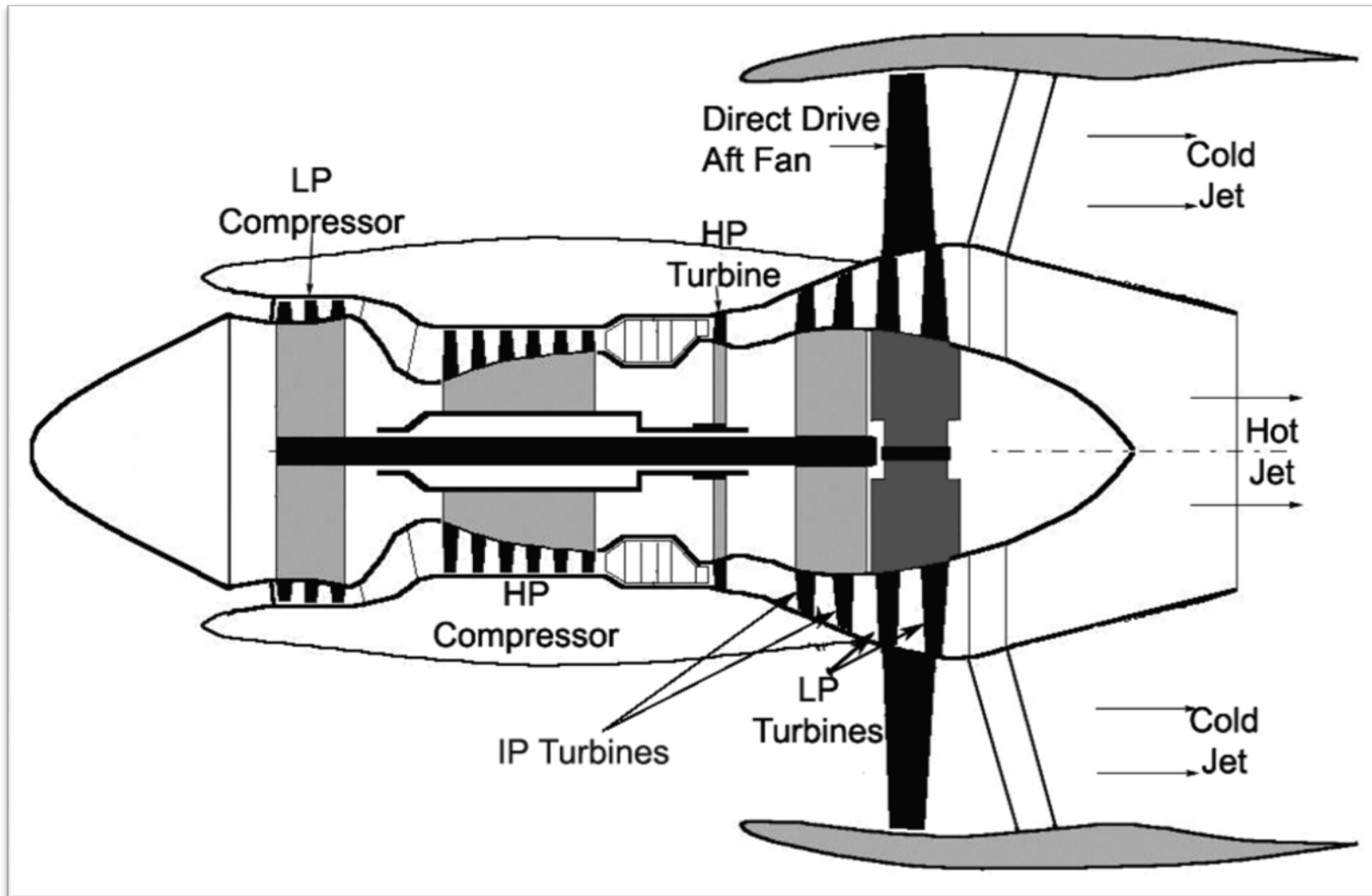


## Three spool geared contra-rotating aft prop-fans





## Two-spool high bypass aft-fan turbofan





**Frontal view of a geared two spool very high bypass turbofan engine**

## A contra-rotating twin rotor aft fan prop-fan test



## Contra-rotating twin rotor aft-prop-fan flight test



Aircraft Engine development over the years have proceeded towards making them more compact (light-weight, small sized, measured in Thrust/weight ratio ) and highly fuel efficient .

Recent research and developments focus on the following issues :

- 1) Energy Audit & search for new fuels
- 2) Chemical Pollution Audit
- 3) Noise Audit
- 4) Infra-red signature audit (for military aircraft)



**Space vehicle  
being lifted by a  
rocket engine**